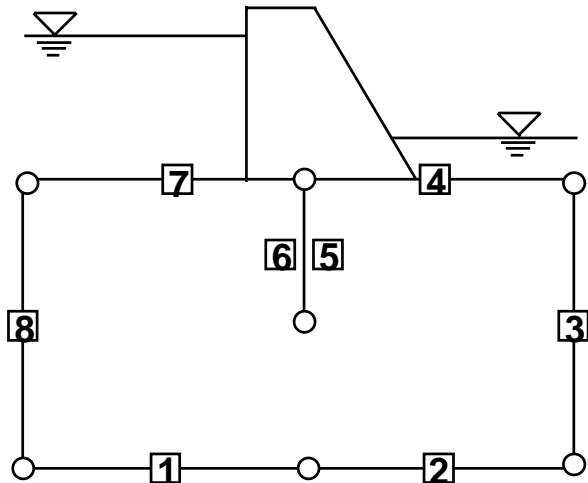
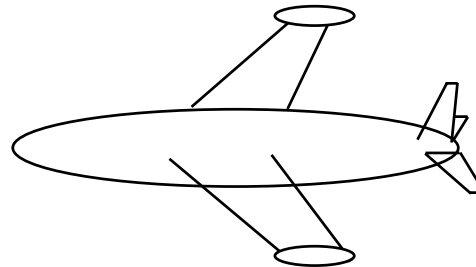


Dual Boundary Element Method and Its Applications

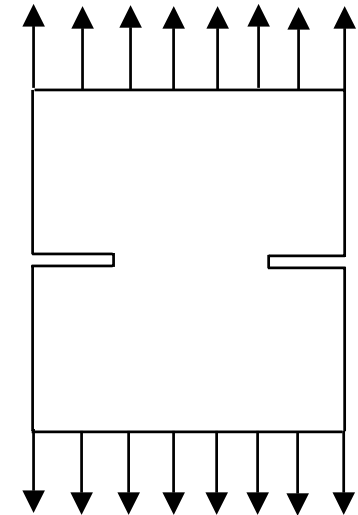
Seepage with
sheetpiles



Thin-airfoill
Aerodynamics



Crack problem



J.T. Chen

Department of Civil Engineering
National Taiwan University

Presentation for Feng Chia University

Apr.,25, 1994



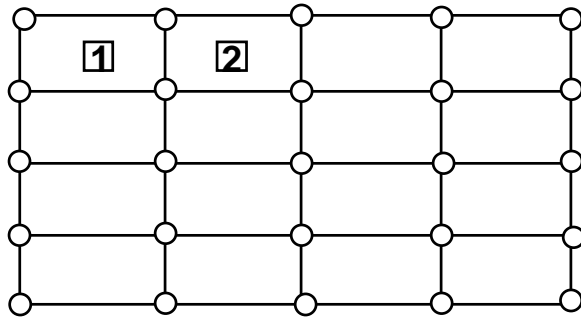
Outline

- **Introduction to BEM**
- **Introduction to dual BEM**
- **Theory of dual integral equations**
- **The role of dual integral equations**
- **Discussion on singular integrals**
- **Applications**
- **Conclusions**



What Is Boundary Element Method ?

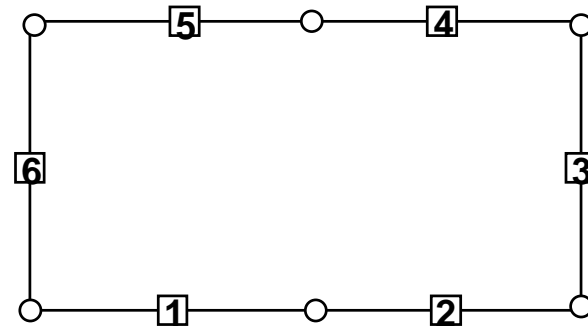
• Finite element method



○ geometry node

American doctor !

Boundary element method

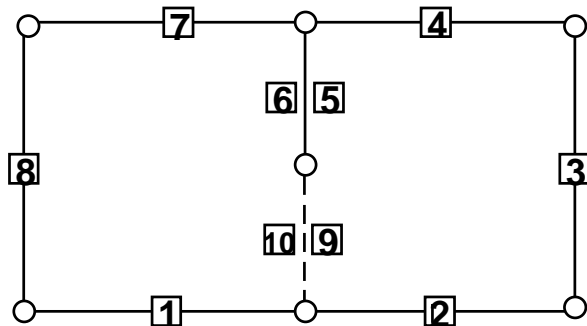


▣ the Nth constant or linear element

Chinese doctor !

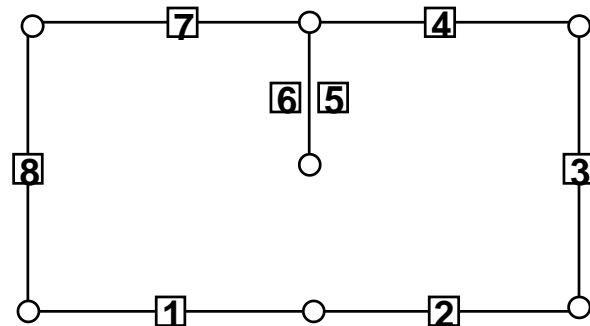
What Is Dual Boundary Element Method ?

- **Boundary element method**



Artificial boundary introduced !

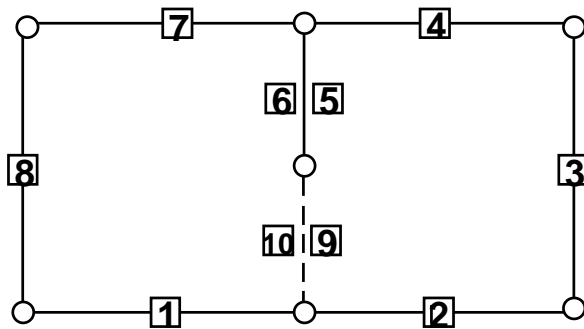
Dual boundary element method



Dual integral equations needed !

Theory of Dual Integral Equations Dual Boundary Element Method

• Boundary element method

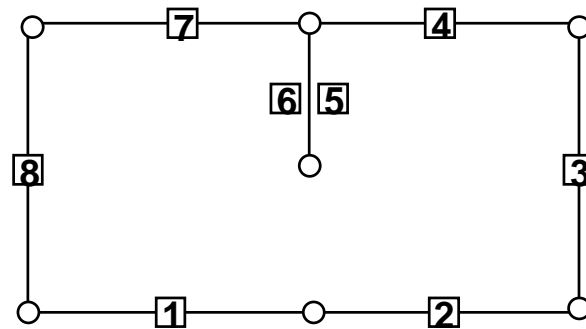


Only U, T equation is used

$$2\pi u(x) = \int_B T(s, x)u(s)dB(s) - \int_B U(s, x)t(s)dB(s), \quad x \in D$$

$$2\pi t(x) = \int_B M(s, x)u(s)dB(s) - \int_B L(s, x)t(s)dB(s), \quad x \in D$$

Dual boundary element method



Both equations are used

Theory of Dual Integral Equations

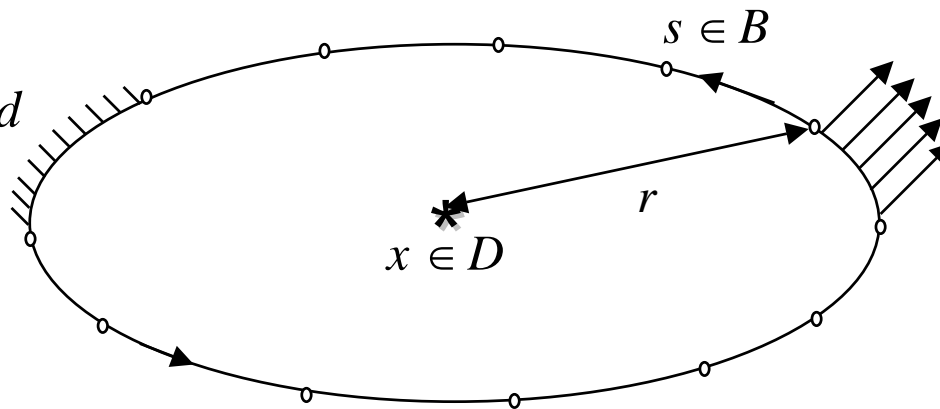
- Dual integral equations for domain point

$$2\pi u(x) = \int_B T(s, x)u(s)dB(s) - \int_B U(s, x)t(s)dB(s), \quad x \in D$$

$$2\pi t(x) = \int_B M(s, x)u(s)dB(s) - \int_B L(s, x)t(s)dB(s), \quad x \in D$$

$u(s) = \text{specified}$

$t(s) = ?$



Theory of Dual Integral Equations

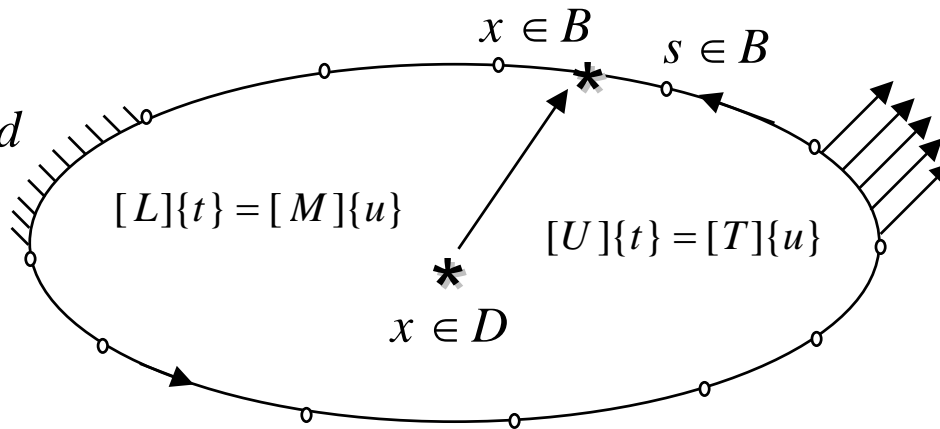
- Dual integral equations for boundary point

$$\pi u(x) = C.P.V. \int_B T(s, x) u(s) dB(s) - R.P.V. \int_B U(s, x) t(s) dB(s), \quad x \in B$$

$$\pi t(x) = H.P.V. \int_B M(s, x) u(s) dB(s) - C.P.V. \int_B L(s, x) t(s) dB(s), \quad x \in B$$

$u(s) = \text{specified}$

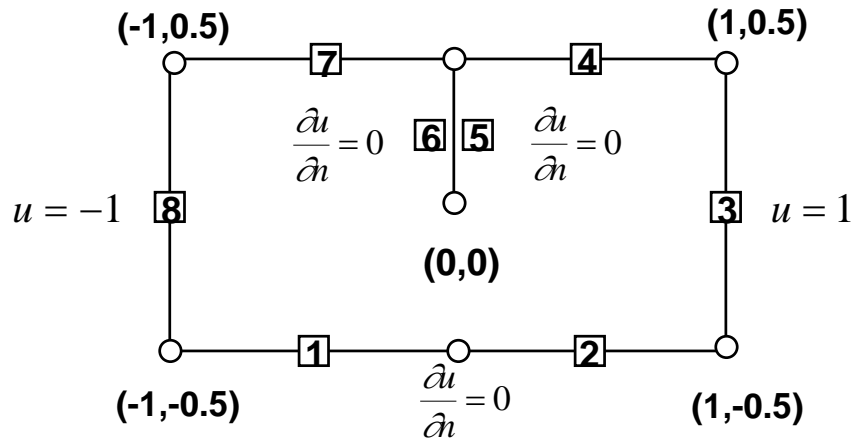
$t(s) = ?$



$t(s) = \text{specified}$

$u(s) = ?$

Degeneracy of the Degenerate Boundary



- geometry node
- ◻ the Nth constant or linear element

$$[U]\{t\} = [T]\{u\}$$

$$[L]\{t\} = [M]\{u\}$$

	$n(s)$							
	5(+)	6(+)						
$[U] =$	1.693	-0.045	0.471	0.347	-0.054	-0.054	0.039	-0.335
	-0.045	-1.693	-0.335	0.039	-0.054	-0.054	0.347	0.471
	0.445	-0.335	-1.693	-0.335	0.019	0.019	0.445	0.703
	0.347	0.039	-0.335	-1.693	-0.281	-0.281	-0.045	0.471
	-0.081	-0.081	0.063	-0.638	-1.193	-1.193	-0.638	0.063
	-0.081	-0.081	0.063	-0.638	-1.193	-1.193	-0.638	0.063
	0.039	0.347	0.471	-0.045	-0.281	-0.281	-1.693	-0.334
	-0.335	0.445	0.703	0.445	0.019	0.019	-0.335	-1.693

	$n(x)$							
	5(+)	6(+)						
$[L] =$	π	0.000	0.184	0.519	0.458	0.458	0.927	0.805
	0.000	π	0.805	0.927	0.458	0.458	0.519	0.184
	0.612	0.805	π	0.805	0.464	0.464	0.612	0.490
	0.519	0.927	0.805	π	0.347	0.347	0.000	0.184
	0.511	0.511	0.888	1.417	π	$-\pi$	-1.417	-0.888
	0.511	0.511	-0.888	-1.417	$-\pi$	π	1.417	0.888
	0.927	0.519	0.184	0.000	0.347	0.347	π	0.805
	0.805	0.612	0.490	0.612	0.464	0.464	0.805	π

	$n(s)$							
	5(+)	6(-)						
$[T] =$	$-\pi$	0.000	0.588	0.519	-0.321	0.321	0.927	1.107
	0.000	$-\pi$	1.107	0.927	0.321	-0.321	0.519	0.588
	0.219	1.107	$-\pi$	1.107	0.464	-0.464	0.219	0.490
	0.519	0.927	1.107	$-\pi$	0.785	-0.785	0.000	0.588
	0.927	0.927	0.888	1.326	$-\pi$	$-\pi$	1.326	0.888
	0.927	0.927	0.888	1.326	$-\pi$	$-\pi$	1.326	0.888
	0.927	0.519	0.588	0.000	-0.7854	0.785	$-\pi$	1.107
	1.107	0.219	0.490	0.219	-0.464	0.464	1.107	$-\pi$

	$n(x)$							
	5(+)	6(-)						
$[M] =$	4.000	-1.333	-0.205	-0.061	0.600	-0.600	-0.800	-1.600
	-1.333	4.000	-1.600	-0.800	-0.600	0.600	-0.061	-0.205
	0.282	-1.600	4.000	-1.600	-0.400	0.400	-0.282	-0.236
	-0.061	-0.800	-1.600	4.000	-1.000	1.000	-1.333	-0.205
	0.853	-0.853	-0.715	-3.765	8.000	-8.000	3.765	0.715
	-0.853	0.853	0.715	3.765	-8.000	8.000	-3.765	-0.715
	-0.800	-0.062	-0.205	-1.333	1.000	-1.000	4.000	-1.600
	-1.600	-0.282	-0.235	-0.282	0.400	-0.400	-1.600	4.000

Definitions of R.P.V., C.P.V. and H.P.V.

- **R.P.V.**

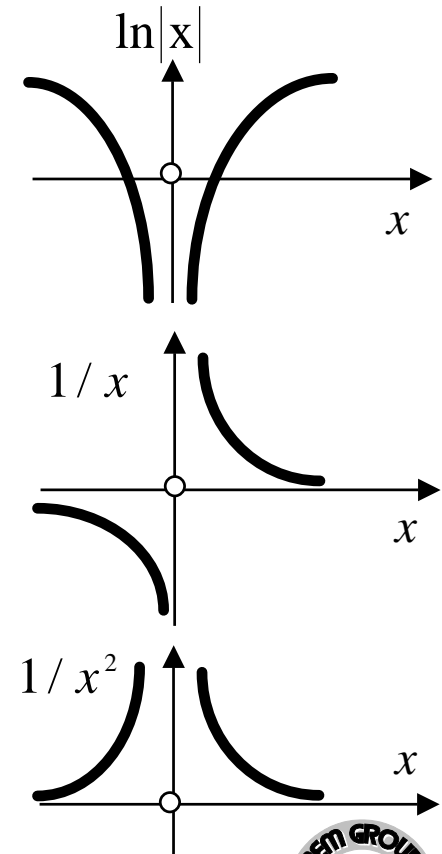
$$R.P.V. \int_{-1}^1 \ln|x| dx = (x \ln|x| - x) \Big|_{x=-1}^{x=1} = -2$$

- **C.P.V.**

$$C.P.V. \int_{-1}^1 \frac{1}{x} dx = \lim_{\varepsilon \rightarrow 0} \int_{-1}^{-\varepsilon} \frac{1}{x} dx + \int_{\varepsilon}^1 \frac{1}{x} dx = 0$$

- **H.P.V.**

$$H.P.V. \int_{-1}^1 \frac{1}{x^2} dx = \lim_{\varepsilon \rightarrow 0} \int_{-1}^{-\varepsilon} \frac{1}{x^2} dx + \int_{\varepsilon}^1 \frac{1}{x^2} dx - \frac{2}{\varepsilon} = -2$$



Roles of hypersingularity in boundary element method

complementary constraints

higher order element

degenerate boundary

corner problem

fictitious eigenvalue

adaptive BEM

secondary field calculation

condition number

symmetry formulation

image system

1. Hermite element

1. cutoff wall
2. sheet pile
3. crack
4. baffle
5. thin airfoil
6. antenna

$$\begin{aligned} (o) \begin{cases} [U](t) = [T](u) \\ [L](t) = [M^{-}](u) \end{cases} \\ (o) \begin{cases} [U](t) = [T](u) \\ [L^{-}](t) = [M^{-}](u) \end{cases} \\ (x) \begin{cases} [L^{-}](t) = [M^{-}](u) \\ [L](t) = [M^{-}](u) \end{cases} \end{aligned}$$

$$\begin{aligned} 1. \begin{cases} [U](t) = [T](u) \\ [L](t) = [M^{-}](u) \end{cases} \\ 2. \begin{cases} [U](t) = [T](u) \\ [L^{-}](t) = [M^{-}](u) \end{cases} \\ 3. \begin{cases} [L^{-}](t) = [M^{-}](u) \\ [L](t) = [M^{-}](u) \end{cases} \end{aligned}$$

1. kernel function
2. region of singularity
3. boundary condition

1. error estimator

1. hoop stress on boundary
2. tangent flux along boundary
3. regularized version for stress near boundary

1. pseudo-differential operator

$U(-1)$	$T(0)$
$L(0)$	$M(1)$

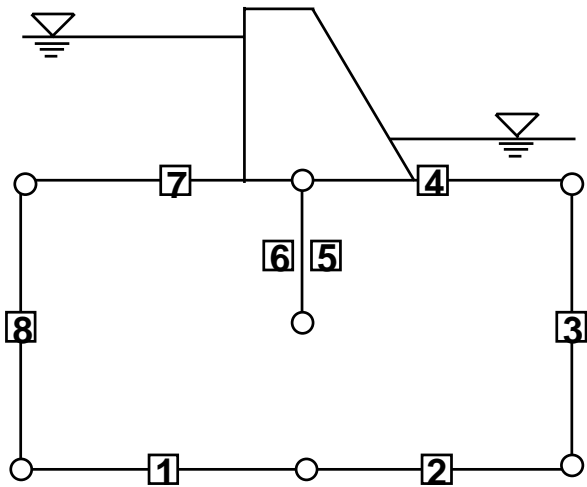
2. T, L is more stable than U, M

1. double boundary integration

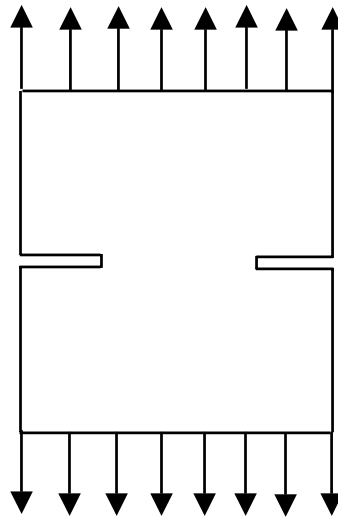
1. normal vector of dipole or dislocation

Applications of Dual Integral Equations

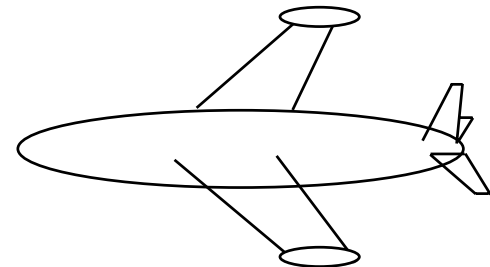
Seepage with sheetpiles



Crack problem



Thin-airfoil Aerodynamics



Conclusions

- The theory of dual integral equation has been introduced
- The role of hypersingularity is examined
- The dual boundary element program has been implemented
- The applications to seepage flow with sheet piles, crack problem and thin airfoil aerodynamics have been demonstrated.

Hypersingularity



Divergent series