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A new method for Stokes problems with circular boundaries using degenerate kernel and Fourier series

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SUMMARY

This study is concerned with the Stokes flow of an incompressible fluid of constant density and viscosity with circular boundaries. To fully capture the circular boundary, the boundary densities in the direct and indirect boundary integral equation (BIE) are expanded in terms of Fourier series. The kernel functions in either the direct BIE or the indirect BIE are expanded to degenerate kernels by using the separation of field and source points. Consequently, the improper integrals are transformed to series sum and are easily calculated. The linear algebraic system can be established by matching the boundary conditions at the collocation points. Then, the unknown Fourier coefficients can be easily determined. Finally, several examples including circular and eccentric domains are presented to demonstrate the validity of the present method. Five gains were obtained: (1) meshless approach; (2) free of boundary-layer effect; (3) singularity free; (4) exponential convergence; and (5) well-posed model. Copyright © 2007 John Wiley & Sons, Ltd.

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1. INTRODUCTION

The boundary element method (BEM) by discretizing the boundary integral equation (BIE) has been extensively applied for engineering problems recently more than domain-type methods, e.g. finite element method (FEM) or finite difference method. It is noted that improper integrals on the boundary should be handled particularly when BEM is used. In the past, many researchers proposed several regularization techniques to deal with the singularity and hypersingularity. To determine the Cauchy principal value and the Hadamard principal value in the singular and hypersingular integrals is a critical issue in BEM/BIE method (BIEM). The technique of the integration by parts

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1 to reduce the order of singularity [1] is an alternative. One order of singularity is shifted to the
2 density function from the kernel. In this paper, instead of using the previous concepts, the kernel
3 function is described in an analytical form on each side across the boundary (interior and exterior)
4 by employing the degenerate kernel since the potential is discontinuous across the boundary.
5 Therefore, degenerate kernel, namely separable kernel, is a vital tool to study the Stokes problems
6 with circular boundaries. Boundary integral formulation is nothing more than the linear algebra
7 once fundamental solutions are expressed by separable kernels. One gain is that this formulation
8 is free of singularity.

9 BIEs for the plate problems were acquired from the Rayleigh–Green identity [2, 3] and the
10 null-field integral equations were derived by collocating the field point outside the domain. The
11 formulation for the plate problems can be applied to study the Stokes flow problem since both
12 displacement and stream function satisfy biharmonic equation. The kernel functions in the present
13 formulation are expanded to degenerate kernels in an analytical series representation by separating
14 the source point and field point and the boundary densities are expressed in terms of Fourier
15 series. It is well known that Fourier series is always incorporated to formulate the solution for
16 problems with circular boundaries [4, 5]. Bird and Steele [4] presented a Fourier series procedure
17 to solve circular plate problems containing multiple circular holes. Also, Mogilevskaya and Crouch
18 [6, 7] presented a method in conjunction with Fourier series for solving problems with randomly
19 distributed circular elastic inclusions with arbitrary properties. Although Fourier series expansions
20 have been employed, it seems that no one has ever introduced the degenerate kernel in BIEs to
21 tackle the problem. Therefore, the BIE in conjunction with degenerate kernel and Fourier series
22 is proposed to solve the Stokes problems with circular boundaries. Two gains are that exponential
23 convergence instead of linear algebraic order can be obtained and mesh generation on the boundary
24 is not required.

25 The Stokes flow problem with circular boundaries is considered since the stream function as
26 well as displacement plate problem satisfies the biharmonic equation. The computation for internal
27 Stokes flow problems for a circle by integral equations was solved analytically [8]. Later, Chen
28 *et al.* revisited this problem and obtained the series solution by using degenerate kernel and
29 Fourier series [2]. A spectral boundary element approach to three-dimensional Stokes flow was
30 proposed by Muldowney and Higdon [9]. A numerical approach for Stokes flow past a particle
31 of arbitrary shape was proposed by Youngren and Acrivos [10]. The flow between eccentric
32 cylinders for the doubly connected problem is focused in this paper. Many papers were published
33 on these problems, some important works are those of Kamal [11], DiPrima and Stuart [12].
34 Ingham and Kelmanson [13], Kelmanson [14] and Wannier [15] also applied the BIE to solve
35 the problems of two-dimensional steady slow flow for the lubrication technology. Although both
36 of the Kelmanson's formulation and the present method are based on the same BIE, the main
37 differences are pointed out here. First, the kernel functions in Kelmanson's paper are fundamental
38 solutions instead of degenerate kernels. It is noted that all the improper integrals are transformed
39 to series sum and are easily calculated when the degenerate kernels are used since the potential
40 across the boundary can be described explicitly in both sides, interior and exterior regions. Second,
41 Fourier expansion for the boundary density is used in this paper instead of linear boundary element
42 scheme [13, 14].

43 The purpose of this paper is to study biharmonic problems with circular boundaries by using
44 direct and indirect BIEs in conjunction with degenerate kernels, Fourier series, vector decompo-
45 sition and the adaptive observer frame. It is very convenient to be able to expand the solution
in an alternative form, each form referring to a different specified coordinate set describing the

1 same solution. In the polar coordinate system, the calculation of potential gradients in the normal
 2 and tangential directions for the non-concentric domain must be taken care of. Therefore, the
 3 technique of vector decomposition is adopted to deal with the problem for the non-concentric
 4 case. It is interesting that Stokes flow problem (not involving the Poisson ratio) can also be
 5 solved by using the present formulation for plate although the Poisson ratio is contained in
 6 the approach. Although the well-known alternative BIE formulations for these problems [16]
 7 have been explored, the indirect BIE as well as the direct BIE in conjunction with degenerate
 8 kernels and Fourier series are both used to solve the Stokes problems. Single- and double-layer
 9 potentials are simultaneously used to construct the indirect BIE. Although the indirect method
 10 cannot provide the null-field integral equation, the compatible relationship of the boundary data
 11 (single- and double-layer fictitious densities) is obtained by moving the domain point in BIE to
 12 the boundary. Special care must be taken in selecting the appropriate expressions (interior and
 13 exterior) for the kernel function. Regarding the direct BIE, we employ the concept of null-field
 14 integral equations and collocate the point on the real boundary in real implementation. Finally,
 15 several examples are presented to show the validity of the present method and some conclusions
 are made.

17 2. FORMULATION OF THE STOKES FLOW PROBLEMS

The governing equation of Stokes flow is derived from the Navier–Stokes equation as follows:

$$19 \quad \rho \frac{D\tilde{V}}{Dt} = \rho g - \nabla P + \mu \nabla^2 \tilde{V} \quad (1)$$

20 where \tilde{V} denotes the velocity field $\tilde{V} = (v_r, v_\theta)$, ρ the density of fluid, t the time, g the gravity, P
 21 the pressure and μ the viscosity. Therefore, the first term of Equation (1) means inertia force, the
 22 second term denotes body force, the third term is pressure gradient and the final term is viscous
 23 force. The term of inertia force can be neglected since the low Reynolds number flow is considered
 24 (inertia force \ll viscous force) and the body force is also neglected to reduce Equation (1) as
 25 follows:

$$\nabla P = \mu \nabla^2 \tilde{V} \quad (2)$$

27 The continuity equation for the incompressible two-dimensional flow is expressed as follows:

$$\frac{1}{r} \frac{\partial(rv_r)}{\partial r} + \frac{1}{r} \frac{\partial v_\theta}{\partial \theta} = 0 \quad (3)$$

29 and the velocity components, v_r and v_θ , can be related to the stream function $u(r, \theta)$ through the
 equations

$$v_r = \frac{1}{r} \frac{\partial u}{\partial \theta} \quad (4)$$

$$31 \quad v_\theta = -\frac{\partial u}{\partial r} \quad (5)$$

1 The biharmonic equation can be derived by associating Equations (2)–(5) as follows:

$$\nabla^4 u = 0 \tag{6}$$

3 Introducing the vorticity as the Laplacian of the stream function u [13, 14], we have

$$\nabla^2 u = \omega \tag{7}$$

$$\nabla^2 \omega = 0 \tag{8}$$

5 where ω is the vorticity. To deal with the Stokes problem, two ways are used in the literature
 6 [3, 14]. First, the biharmonic equation of Equation (6) is treated [3]. The other one is solving the
 7 Poisson and Laplace equation in Equations (7)–(8) [14]. In this paper, we focus on the former
 8 formulation.

3. DIRECT BIE METHOD

9 3.1. BIE for the domain point

10 Here, we use plate formulation to solve Stokes problems since they both satisfy the biharmonic
 11 equation. The direct BIEs for the domain point can be derived from the Rayleigh–Green identity
 [2, 3] as follows:

$$\begin{aligned} 8\pi u(x) = & - \int_B U(s, x)v(s) dB(s) + \int_B \Theta(s, x)m(s) dB(s) \\ & - \int_B M(s, x)\theta(s) dB(s) + \int_B V(s, x)u(s) dB(s), \quad x \in \Omega \end{aligned} \tag{9}$$

$$\begin{aligned} 8\pi \theta(x) = & - \int_B U_\theta(s, x)v(s) dB(s) + \int_B \Theta_\theta(s, x)m(s) dB(s) \\ & - \int_B M_\theta(s, x)\theta(s) dB(s) + \int_B V_\theta(s, x)u(s) dB(s), \quad x \in \Omega \end{aligned} \tag{10}$$

$$\begin{aligned} 8\pi m(x) = & - \int_B U_m(s, x)v(s) dB(s) + \int_B \Theta_m(s, x)m(s) dB(s) \\ & - \int_B M_m(s, x)\theta(s) dB(s) + \int_B V_m(s, x)u(s) dB(s), \quad x \in \Omega \end{aligned} \tag{11}$$

$$\begin{aligned} 8\pi v(x) = & - \int_B U_v(s, x)v(s) dB(s) + \int_B \Theta_v(s, x)m(s) dB(s) \\ & - \int_B M_v(s, x)\theta(s) dB(s) + \int_B V_v(s, x)u(s) dB(s), \quad x \in \Omega \end{aligned} \tag{12}$$

13

1 where B is the boundary of the domain Ω , $u(x)$, $\theta(x)$, $m(x)$ and $v(x)$ are the displacement, slope,
 3 moment and shear force for solid mechanics, s and x are the source point and field point, respec-
 tively. However, $u(x)$ is defined as the stream function in this paper instead of displacement
 for plate problem. The kernel functions U , Θ , M , V , U_θ , Θ_θ , M_θ , V_θ , U_m , Θ_m , M_m , V_m , U_v , Θ_v ,
 5 M_v , V_v in Equations (9)–(12) are expanded to degenerate kernels by using the separation of source
 and field points [3, 17]. The kernel function $U(s, x)$ in Equation (9) is the fundamental solution
 7 that satisfies

$$\nabla^4 U(s, x) = 8\pi\delta(s - x) \quad (13)$$

9 where $\delta(s - x)$ is the Dirac-delta function. Therefore, the fundamental solution can be obtained

$$U(s, x) = r^2 \ln r \quad (14)$$

11 where r is the distance between source point s and field point x . The relationship among
 $u(x)$, $\theta(x)$, $m(x)$ and $v(x)$ are shown as follows:

$$\theta(x) = K_{\theta,x}(u(x)) = \frac{\partial u(x)}{\partial n_x} \quad (15)$$

$$m(x) = K_{m,x}(u(x)) = v\nabla_x^2 u(x) + (1-v)\frac{\partial^2 u(x)}{\partial^2 n_x} \quad (16)$$

$$v(x) = K_{v,x}(u(x)) = \frac{\partial \nabla_x^2 u(x)}{\partial n_x} + (1-v)\frac{\partial}{\partial t_x} \left[\frac{\partial}{\partial n_x} \left(\frac{\partial u(x)}{\partial t_x} \right) \right] \quad (17)$$

13 where $K_{\theta,x}(\cdot)$, $K_{m,x}(\cdot)$, $K_{v,x}(\cdot)$ are the slope, moment and shear force operators with respect to
 the point x , $\partial/\partial n_x$ is the normal derivative with respect to the field point x , $\partial/\partial t_x$ is the tangential
 15 derivative with respect to the field point x , ∇_x^2 means the Laplacian operator and v is the Poisson
 ratio.

17 By taking the Laplacian with respect to $u(x)$ in Equation (9), the vorticity function is derived
 as follows:

$$\begin{aligned} 8\pi\omega(x) = & - \int_B U_{\nabla^2}(s, x)v(s) dB(s) + \int_B \Theta_{\nabla^2}(s, x)v(s) dB(s) \\ & - \int_B M_{\nabla^2}(s, x)v(s) dB(s) + \int_B V_{\nabla^2}(s, x)v(s) dB(s), \quad x \in \Omega \end{aligned} \quad (18)$$

19 where $U_{\nabla^2}(s, x)$, $\Theta_{\nabla^2}(s, x)$, $M_{\nabla^2}(s, x)$ and $V_{\nabla^2}(s, x)$ are the Laplacian of degenerate kernels
 $U(s, x)$, $\Theta(s, x)$, $M(s, x)$ and $V(s, x)$, respectively. The kernel functions are listed in Appendix A.

21 By using the formulations in conjunction with the degenerate kernels, Fourier series and adaptive
 observer system, the stream function and vorticity can be solved.

1 *3.2. Null-field integral equation*

3 The null-field integral equations were obtained by collocating the field point x outside the domain as follows:

$$\begin{aligned}
 0 = & - \int_B U(s, x)v(s) dB(s) + \int_B \Theta(s, x)m(s) dB(s) \\
 & - \int_B M(s, x)\theta(s) dB(s) + \int_B V(s, x)u(s) dB(s), \quad x \in \Omega^C
 \end{aligned} \tag{19}$$

$$\begin{aligned}
 0 = & - \int_B U_\theta(s, x)v(s) dB(s) + \int_B \Theta_\theta(s, x)m(s) dB(s) \\
 & - \int_B M_\theta(s, x)\theta(s) dB(s) + \int_B V_\theta(s, x)u(s) dB(s), \quad x \in \Omega^C
 \end{aligned} \tag{20}$$

$$\begin{aligned}
 0 = & - \int_B U_m(s, x)v(s) dB(s) + \int_B \Theta_m(s, x)m(s) dB(s) \\
 & - \int_B M_m(s, x)\theta(s) dB(s) + \int_B V_m(s, x)u(s) dB(s), \quad x \in \Omega^C
 \end{aligned} \tag{21}$$

$$\begin{aligned}
 0 = & - \int_B U_v(s, x)v(s) dB(s) + \int_B \Theta_v(s, x)m(s) dB(s) \\
 & - \int_B M_v(s, x)\theta(s) dB(s) + \int_B V_v(s, x)u(s) dB(s), \quad x \in \Omega^C
 \end{aligned} \tag{22}$$

5 where Ω^C is the complementary domain of Ω . Since the four equations of Equations (19)–(22)
 7 are given, there are six (C_2^4) options for choosing any two equations to solve the problems. For
 9 simplicity, Equations (19) and (20) are used. In the real implementation, the collocation point in the
 null-field integral equation is moved to the boundary from Ω^C such that the kernel functions can
 be expressed in terms of appropriate forms of degenerate kernels. Consequently, all the improper
 integrals disappear and are transformed to series sum in the BIEs since the potential across the
 boundary can be described explicitly in both sides by using degenerate kernels.

11 *3.3. Expansion of Fourier series*

The boundary densities $u(s)$, $\theta(s)$, $m(s)$ and $v(s)$ are expressed in terms of Fourier series as follows:

$$u(s) = p_0 + \sum_{n=1}^M (p_n \cos n\theta + q_n \sin n\theta) \tag{23}$$

$$\theta(s) = g_0 + \sum_{n=1}^M (g_n \cos n\theta + h_n \sin n\theta) \tag{24}$$

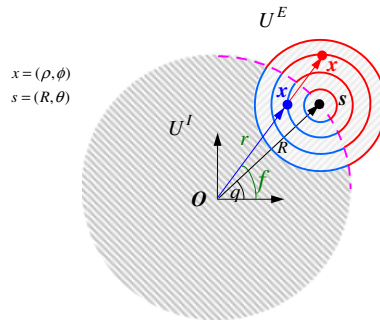


Figure 1. Degenerate kernel for $U(s, x)$.

$$m(s) = c_0 + \sum_{n=1}^M (c_n \cos n\theta + d_n \sin n\theta) \tag{25}$$

$$u(s) = a_0 + \sum_{n=1}^M (a_n \cos n\theta + b_n \sin n\theta) \tag{26}$$

1 where $a_0, a_n, b_n, c_0, c_n, d_n, g_0, g_n, h_n, p_0, p_n$ and q_n are Fourier coefficients and M denotes the truncating terms of Fourier series.

3 **3.4. Expansion of kernels**

5 By employing the separation technique for source and field points, the kernel function $U(s, x)$ can be expanded in terms of degenerate kernel in a series form [17] as shown below:

$$U(s, x) = r^2 \ln r$$

$$= \begin{cases} U^I(s, x) = \rho^2(1 + \ln R) + R^2 \ln R - \left[R\rho(1 + 2 \ln R) + \frac{1}{2} \frac{\rho^3}{R} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{1}{m(m+1)} \frac{\rho^{m+2}}{R^m} - \frac{1}{m(m-1)} \frac{\rho^m}{R^{m-2}} \right] \cos[m(\theta - \phi)], \quad R \geq \rho \\ U^E(s, x) = R^2(1 + \ln \rho) + \rho^2 \ln \rho - \left[\rho R(1 + 2 \ln \rho) + \frac{1}{2} \frac{R^3}{\rho} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{1}{m(m+1)} \frac{R^{m+2}}{\rho^m} - \frac{1}{m(m-1)} \frac{R^m}{\rho^{m-2}} \right] \cos[m(\theta - \phi)], \quad \rho > R \end{cases} \tag{27a, 27b}$$

7 where the superscripts 'I' and 'E' denote the interior and exterior cases of $U(s, x)$ kernel depending
 9 on the geometry as shown in Figure 1. It is interesting to find that interior and exterior Trefftz
 11 bases are imbedded in the degenerate kernel. The other kernels in the BIEs can be obtained by
 utilizing the operators of Equations (15)–(17) with respect to the $U(s, x)$ kernel. The degenerate
 kernels $U, \Theta, M, V, U_\theta, \Theta_\theta, M_\theta$ and V_θ in Equations (9) and (10) are listed in Appendix A. It is

1 noted that the interior and exterior cases of U, Θ, M, U_θ and Θ_θ are the same when they both
 2 approach the boundary ($\rho = R$), since the degenerate kernels are continuous functions across the
 3 boundary. Then, the kernel function with the superscript 'I' is chosen while the field point is inside
 the circular region; otherwise, the kernels with the superscript 'E' are chosen.

5 4. INDIRECT BIE METHOD

Indirect BIE method is originated from the physical concept of superposition and must satisfy not
 7 only the governing equation but also the boundary conditions. There are four kinds of potentials,
 single-, double-, triple- and quadruple-layer potentials in the indirect BIEM for the Stokes flow
 9 problems. By choosing any two potentials, six options (C_2^4) (single–double-layer potentials, single–
 triple-layer potentials, single–quadruple-layer potentials, double–triple-layer potentials, double–
 11 quadruple-layer potentials and triple–quadruple-layer potentials) can be chosen. For simplicity,
 single- and double-layer potentials are chosen here as follows:

13
$$u(x) = \int_B U(s, x)\Phi(s) dB(s) + \int_B \Theta(s, x)\Psi(s) dB(s), \quad x \in \Omega \tag{28}$$

where $\Phi(s)$ and $\Psi(s)$ are the single- and double-layer fictitious densities, respectively, and B is
 15 the boundary of the domain Ω . By taking normal derivative with respect to $u(x)$ in Equation (28),
 we have

17
$$\theta(x) = \int_B U_\theta(s, x)\Phi(s) dB(s) + \int_B \Theta_\theta(s, x)\Psi(s) dB(s), \quad x \in \Omega \tag{29}$$

19 The single- and double-layer fictitious densities in Equations (28) and (29) are expressed in terms
 of Fourier series as follows:

$$\Phi(s) = a_0 + \sum_{n=1}^M (a_n \cos n\theta + b_n \sin n\theta) \tag{30}$$

$$\Psi(s) = c_0 + \sum_{n=1}^M (c_n \cos n\theta + d_n \sin n\theta) \tag{31}$$

21 where a_0, a_n, b_n, c_0, c_n and d_n are the Fourier coefficients and M denotes the truncating terms of
 Fourier series. By taking the Laplacian with respect to $u(x)$ in Equation (28), the vorticity function
 is derived as shown below:

23
$$\omega(x) = \int_B U_{\nabla^2}(s, x)\Phi(s) dB(s) + \int_B \Theta_{\nabla^2}(s, x)\Psi(s) dB(s), \quad x \in \Omega \tag{32}$$

25 where $U_{\nabla^2}(s, x)$ and $\Theta_{\nabla^2}(s, x)$ are the Laplacian of the degenerate kernels $U(s, x)$ and $\Theta(s, x)$,
 respectively. It is noted that null-field integral equation in the indirect method is not available.
 However, the compatible relationship of boundary data can be obtained by moving the domain
 27 point x in Equations (28) and (29) to the boundary B^- and B^+ from inside and outside domains,
 respectively.

1 5. ADAPTIVE OBSERVER SYSTEM AND VECTOR DECOMPOSITION FOR THE
 NORMAL DERIVATIVE

3 5.1. Adaptive observer system

4 Consider a biharmonic problem with circular boundaries as shown in Figure 2. Since the BIEs
 5 are frame indifferent due to objectivity, an adaptive observer system is chosen to fully employ the
 6 circular property by expanding the kernels into degenerate forms. The origin of the observer system
 7 can be adaptively located on the center of the corresponding boundary contour under integration.
 8 The dummy variable in the circular contour integration is the angle (θ) instead of radial coordinate
 9 (R). By using the adaptive system, all the boundary integrals can be determined analytically free
 of principal value senses.

11 5.2. Vector decomposition

12 Since the higher-order singular equation is also one alternative to deal with the Stokes problem,
 13 potential gradient or higher-order gradients is required to calculate carefully. For the non-concentric
 14 case, special treatment for the potential gradient should be taken care as the source and field points
 15 locate on different boundaries. As shown in Figure 3, the true normal direction with respect to
 the collocation point x on the B_j boundary can be superimposed by using the radial direction \underline{e}_ρ
 17 and angular direction \underline{e}_ϕ on the B_j boundary. The degenerate kernels for the higher-order singular
 equation (θ -formulation) are changed to

$$U_n(s, x) = \frac{\partial U(s, x)}{\partial n_x} \cos(\phi - \phi') + \frac{\partial U(s, x)}{\partial t_x} \cos\left(\frac{\pi}{2} - \phi + \phi'\right) \quad (33)$$

$$\Theta_n(s, x) = \frac{\partial \Theta(s, x)}{\partial n_x} \cos(\phi - \phi') + \frac{\partial \Theta(s, x)}{\partial t_x} \cos\left(\frac{\pi}{2} - \phi + \phi'\right) \quad (34)$$

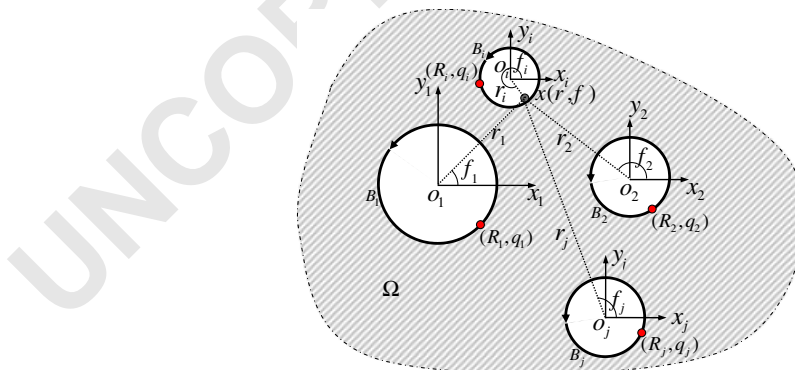


Figure 2. Adaptive observer system at O_j ($j = 1, 2, 3, \dots, N_c$) when integrating the corresponding circular boundary B_j for the collocation null-field point near B_i .

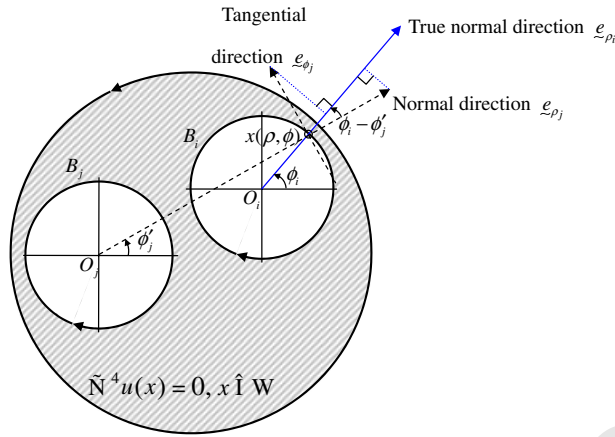


Figure 3. Vector decomposition (Collocation on x and integration on B_j).

$$M_n(s, x) = \frac{\partial M(s, x)}{\partial n_x} \cos(\phi - \phi') + \frac{\partial M(s, x)}{\partial t_x} \cos\left(\frac{\pi}{2} - \phi + \phi'\right) \quad (35)$$

$$V_n(s, x) = \frac{\partial V(s, x)}{\partial n_x} \cos(\phi - \phi') + \frac{\partial V(s, x)}{\partial t_x} \cos\left(\frac{\pi}{2} - \phi + \phi'\right) \quad (36)$$

- 1 The tangential derivative $\partial/\partial t_x$ with respect to the field point x for the four kernels need to be
 3 additionally derived and are listed in Appendix A, where the normal derivative $\partial/\partial n_x$ is $\partial/\partial \rho$ and
 5 has been derived in the $U_\theta, \Theta_\theta, M_\theta$ and V_θ kernels. We call this treatment ‘vector decomposition
 technique’. By approaching the collocation point from Ω^C to B_i and integrating the B_j circle using
 the adaptive observer system of origin O_j , the normal and tangent derivatives can be superimposed
 as follows:

$$\frac{\partial}{\partial \rho_i} = \frac{\partial}{\partial \rho_j} \cos(\phi_i - \phi_j') + \frac{1}{\rho_j} \frac{\partial}{\partial \phi_j} \cos\left(\frac{\pi}{2} - \phi_i + \phi_j'\right) \quad (37)$$

$$\frac{1}{\rho_i} \frac{\partial}{\partial \phi_i} = \frac{\partial}{\partial \rho_j} \cos\left(\frac{\pi}{2} - \phi_i + \phi_j'\right) + \frac{1}{\rho_j} \frac{\partial}{\partial \phi_j} \cos(\phi_i - \phi_j') \quad (38)$$

7

6. SOLUTION PROCEDURES OF THE SEMI-ANALYTICAL APPROACHES

- 9 Two semi-analytical approaches, the direct and indirect BIEMs are described. Direct BIEM employs
 the concept of the null-field integral equation but collocates on the real boundary and the indirect
 11 BIEM obtains the compatible relation of boundary data by collocating the point to the boundary
 from the BIE of domain point.

1 *6.1. Direct formulation*

3 *6.1.1. Eccentric case (doubly connected domain).* By using the null-field integral equations (19)–
 3 (20) as shown in Figures 4 and 5, the linear algebraic system can be constructed as follows:

$$\begin{bmatrix} \mathbf{U}_{11} & \Theta_{11} & \mathbf{U}_{12} & \Theta_{12} \\ \mathbf{U}_{11\theta} & \Theta_{11\theta} & \mathbf{U}_{12\theta} & \Theta_{12\theta} \\ \mathbf{U}_{21} & \Theta_{21} & \mathbf{U}_{22} & \Theta_{22} \\ \mathbf{U}_{21\theta} & \Theta_{21\theta} & \mathbf{U}_{22\theta} & \Theta_{22\theta} \end{bmatrix} \begin{Bmatrix} \mathbf{v}_1 \\ \mathbf{m}_1 \\ \mathbf{v}_2 \\ \mathbf{m}_2 \end{Bmatrix} = \begin{bmatrix} \mathbf{M}_{11} & \mathbf{V}_{11} & \mathbf{M}_{12} & \mathbf{V}_{12} \\ \mathbf{M}_{11\theta} & \mathbf{V}_{11\theta} & \mathbf{M}_{12\theta} & \mathbf{V}_{12\theta} \\ \mathbf{M}_{21} & \mathbf{V}_{21} & \mathbf{M}_{22} & \mathbf{V}_{22} \\ \mathbf{M}_{21\theta} & \mathbf{V}_{21\theta} & \mathbf{M}_{22\theta} & \mathbf{V}_{22\theta} \end{bmatrix} \begin{Bmatrix} \theta_1 \\ \mathbf{u}_1 \\ \theta_2 \\ \mathbf{u}_2 \end{Bmatrix} \quad (39)$$

5 For brevity, a unified form $[\mathbf{U}_{ij}]$ ($i=1,2$ and $j=1,2$) denotes the response of $U(s,x)$
 7 kernel at the i th circle point due to the source at the j th circle. Otherwise, the same
 7 definition for $[\Theta_{ij}]$, $[\mathbf{M}_{ij}]$, $[\mathbf{V}_{ij}]$, $[\mathbf{U}_{ij\theta}]$, $[\Theta_{ij\theta}]$, $[\mathbf{M}_{ij\theta}]$ and $[\mathbf{V}_{ij\theta}]$ cases. The sub-matrices
 $[\mathbf{U}_{ij}]$, $[\Theta_{ij}]$, $[\mathbf{M}_{ij}]$, $[\mathbf{V}_{ij}]$, $[\mathbf{U}_{ij\theta}]$, $[\Theta_{ij\theta}]$, $[\mathbf{M}_{ij\theta}]$ and $[\mathbf{V}_{ij\theta}]$ are defined as follows:

$$[\Theta_{ij}] = \begin{bmatrix} \Theta_{ij}^{0c}(\phi_1) & \Theta_{ij}^{1c}(\phi_1) & \Theta_{ij}^{1s}(\phi_1) & \cdots & \Theta_{ij}^{Mc}(\phi_1) & \Theta_{ij}^{Ms}(\phi_1) \\ \Theta_{ij}^{0c}(\phi_2) & \Theta_{ij}^{1c}(\phi_2) & \Theta_{ij}^{1s}(\phi_2) & \cdots & \Theta_{ij}^{Mc}(\phi_2) & \Theta_{ij}^{Ms}(\phi_2) \\ \Theta_{ij}^{0c}(\phi_3) & \Theta_{ij}^{1c}(\phi_3) & \Theta_{ij}^{1s}(\phi_3) & \cdots & \Theta_{ij}^{Mc}(\phi_3) & \Theta_{ij}^{Ms}(\phi_3) \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ \Theta_{ij}^{0c}(\phi_{2M}) & \Theta_{ij}^{1c}(\phi_{2M}) & \Theta_{ij}^{1s}(\phi_{2M}) & \cdots & \Theta_{ij}^{Mc}(\phi_{2M}) & \Theta_{ij}^{Ms}(\phi_{2M}) \\ \Theta_{ij}^{0c}(\phi_{2M+1}) & \Theta_{ij}^{1c}(\phi_{2M+1}) & \Theta_{ij}^{1s}(\phi_{2M+1}) & \cdots & \Theta_{ij}^{Mc}(\phi_{2M+1}) & \Theta_{ij}^{Ms}(\phi_{2M+1}) \end{bmatrix} \quad (40)$$

$$[\mathbf{M}_{ij}] = \begin{bmatrix} M_{ij}^{0c}(\phi_1) & M_{ij}^{1c}(\phi_1) & M_{ij}^{1s}(\phi_1) & \cdots & M_{ij}^{Mc}(\phi_1) & M_{ij}^{Ms}(\phi_1) \\ M_{ij}^{0c}(\phi_2) & M_{ij}^{1c}(\phi_2) & M_{ij}^{1s}(\phi_2) & \cdots & M_{ij}^{Mc}(\phi_2) & M_{ij}^{Ms}(\phi_2) \\ M_{ij}^{0c}(\phi_3) & M_{ij}^{1c}(\phi_3) & M_{ij}^{1s}(\phi_3) & \cdots & M_{ij}^{Mc}(\phi_3) & M_{ij}^{Ms}(\phi_3) \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ M_{ij}^{0c}(\phi_{2M}) & M_{ij}^{1c}(\phi_{2M}) & M_{ij}^{1s}(\phi_{2M}) & \cdots & M_{ij}^{Mc}(\phi_{2M}) & M_{ij}^{Ms}(\phi_{2M}) \\ M_{ij}^{0c}(\phi_{2M+1}) & M_{ij}^{1c}(\phi_{2M+1}) & M_{ij}^{1s}(\phi_{2M+1}) & \cdots & M_{ij}^{Mc}(\phi_{2M+1}) & M_{ij}^{Ms}(\phi_{2M+1}) \end{bmatrix} \quad (41)$$

$$[\mathbf{V}_{ij}] = \begin{bmatrix} V_{ij}^{0c}(\phi_1) & V_{ij}^{1c}(\phi_1) & V_{ij}^{1s}(\phi_1) & \cdots & V_{ij}^{Mc}(\phi_1) & V_{ij}^{Ms}(\phi_1) \\ V_{ij}^{0c}(\phi_2) & V_{ij}^{1c}(\phi_2) & V_{ij}^{1s}(\phi_2) & \cdots & V_{ij}^{Mc}(\phi_2) & V_{ij}^{Ms}(\phi_2) \\ V_{ij}^{0c}(\phi_3) & V_{ij}^{1c}(\phi_3) & V_{ij}^{1s}(\phi_3) & \cdots & V_{ij}^{Mc}(\phi_3) & V_{ij}^{Ms}(\phi_3) \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ V_{ij}^{0c}(\phi_{2M}) & V_{ij}^{1c}(\phi_{2M}) & V_{ij}^{1s}(\phi_{2M}) & \cdots & V_{ij}^{Mc}(\phi_{2M}) & V_{ij}^{Ms}(\phi_{2M}) \\ V_{ij}^{0c}(\phi_{2M+1}) & V_{ij}^{1c}(\phi_{2M+1}) & V_{ij}^{1s}(\phi_{2M+1}) & \cdots & V_{ij}^{Mc}(\phi_{2M+1}) & V_{ij}^{Ms}(\phi_{2M+1}) \end{bmatrix} \quad (42)$$

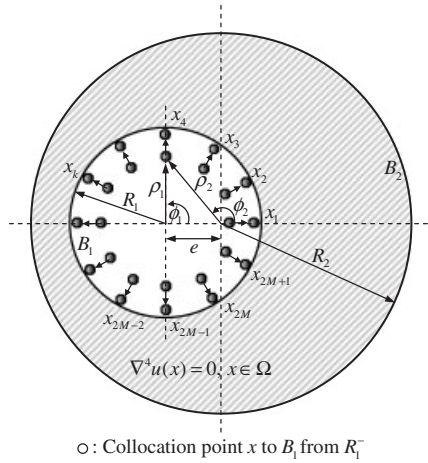


Figure 4. Sketch of the null-field points near the inner cylinder for the centric case.

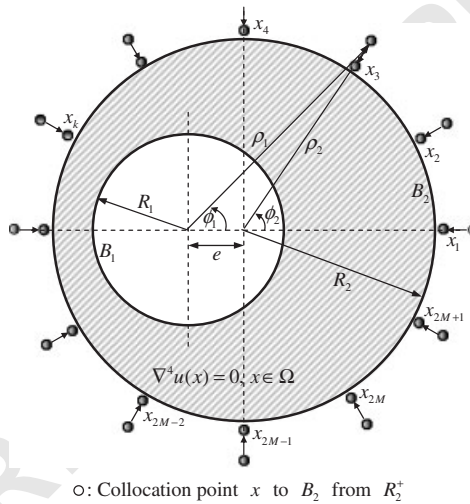


Figure 5. Sketch of the null-field points near the outer cylinder for the eccentric case.

$$[U_{ij\theta}] = \begin{bmatrix}
 U_{ij\theta}^{0c}(\phi_1) & U_{ij\theta}^{1c}(\phi_1) & U_{ij\theta}^{1s}(\phi_1) & \cdots & U_{ij\theta}^{Mc}(\phi_1) & U_{ij\theta}^{Ms}(\phi_1) \\
 U_{ij\theta}^{0c}(\phi_2) & U_{ij\theta}^{1c}(\phi_2) & U_{ij\theta}^{1s}(\phi_2) & \cdots & U_{ij\theta}^{Mc}(\phi_2) & U_{ij\theta}^{Ms}(\phi_2) \\
 U_{ij\theta}^{0c}(\phi_3) & U_{ij\theta}^{1c}(\phi_3) & U_{ij\theta}^{1s}(\phi_3) & \cdots & U_{ij\theta}^{Mc}(\phi_3) & U_{ij\theta}^{Ms}(\phi_3) \\
 \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\
 U_{ij\theta}^{0c}(\phi_{2M}) & U_{ij\theta}^{1c}(\phi_{2M}) & U_{ij\theta}^{1s}(\phi_{2M}) & \cdots & U_{ij\theta}^{Mc}(\phi_{2M}) & U_{ij\theta}^{Ms}(\phi_{2M}) \\
 U_{ij\theta}^{0c}(\phi_{2M+1}) & U_{ij\theta}^{1c}(\phi_{2M+1}) & U_{ij\theta}^{1s}(\phi_{2M+1}) & \cdots & U_{ij\theta}^{Mc}(\phi_{2M+1}) & U_{ij\theta}^{Ms}(\phi_{2M+1})
 \end{bmatrix} \quad (43)$$

$$[\Theta_{ij\theta}] = \begin{bmatrix} \Theta_{ij\theta}^{0c}(\phi_1) & \Theta_{ij\theta}^{1c}(\phi_1) & \Theta_{ij\theta}^{1s}(\phi_1) & \cdots & \Theta_{ij\theta}^{Mc}(\phi_1) & \Theta_{ij\theta}^{Ms}(\phi_1) \\ \Theta_{ij\theta}^{0c}(\phi_2) & \Theta_{ij\theta}^{1c}(\phi_2) & \Theta_{ij\theta}^{1s}(\phi_2) & \cdots & \Theta_{ij\theta}^{Mc}(\phi_2) & \Theta_{ij\theta}^{Ms}(\phi_2) \\ \Theta_{ij\theta}^{0c}(\phi_3) & \Theta_{ij\theta}^{1c}(\phi_3) & \Theta_{ij\theta}^{1s}(\phi_3) & \cdots & \Theta_{ij\theta}^{Mc}(\phi_3) & \Theta_{ij\theta}^{Ms}(\phi_3) \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ \Theta_{ij\theta}^{0c}(\phi_{2M}) & \Theta_{ij\theta}^{1c}(\phi_{2M}) & \Theta_{ij\theta}^{1s}(\phi_{2M}) & \cdots & \Theta_{ij\theta}^{Mc}(\phi_{2M}) & \Theta_{ij\theta}^{Ms}(\phi_{2M}) \\ \Theta_{ij\theta}^{0c}(\phi_{2M+1}) & \Theta_{ij\theta}^{1c}(\phi_{2M+1}) & \Theta_{ij\theta}^{1s}(\phi_{2M+1}) & \cdots & \Theta_{ij\theta}^{Mc}(\phi_{2M+1}) & \Theta_{ij\theta}^{Ms}(\phi_{2M+1}) \end{bmatrix} \quad (44)$$

$$[M_{ij\theta}] = \begin{bmatrix} M_{ij\theta}^{0c}(\phi_1) & M_{ij\theta}^{1c}(\phi_1) & M_{ij\theta}^{1s}(\phi_1) & \cdots & M_{ij\theta}^{Mc}(\phi_1) & M_{ij\theta}^{Ms}(\phi_1) \\ M_{ij\theta}^{0c}(\phi_2) & M_{ij\theta}^{1c}(\phi_2) & M_{ij\theta}^{1s}(\phi_2) & \cdots & M_{ij\theta}^{Mc}(\phi_2) & M_{ij\theta}^{Ms}(\phi_2) \\ M_{ij\theta}^{0c}(\phi_3) & M_{ij\theta}^{1c}(\phi_3) & M_{ij\theta}^{1s}(\phi_3) & \cdots & M_{ij\theta}^{Mc}(\phi_3) & M_{ij\theta}^{Ms}(\phi_3) \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ M_{ij\theta}^{0c}(\phi_{2M}) & M_{ij\theta}^{1c}(\phi_{2M}) & M_{ij\theta}^{1s}(\phi_{2M}) & \cdots & M_{ij\theta}^{Mc}(\phi_{2M}) & M_{ij\theta}^{Ms}(\phi_{2M}) \\ M_{ij\theta}^{0c}(\phi_{2M+1}) & M_{ij\theta}^{1c}(\phi_{2M+1}) & M_{ij\theta}^{1s}(\phi_{2M+1}) & \cdots & M_{ij\theta}^{Mc}(\phi_{2M+1}) & M_{ij\theta}^{Ms}(\phi_{2M+1}) \end{bmatrix} \quad (45)$$

$$[V_{ij\theta}] = \begin{bmatrix} V_{ij\theta}^{0c}(\phi_1) & V_{ij\theta}^{1c}(\phi_1) & V_{ij\theta}^{1s}(\phi_1) & \cdots & V_{ij\theta}^{Mc}(\phi_1) & V_{ij\theta}^{Ms}(\phi_1) \\ V_{ij\theta}^{0c}(\phi_2) & V_{ij\theta}^{1c}(\phi_2) & V_{ij\theta}^{1s}(\phi_2) & \cdots & V_{ij\theta}^{Mc}(\phi_2) & V_{ij\theta}^{Ms}(\phi_2) \\ V_{ij\theta}^{0c}(\phi_3) & V_{ij\theta}^{1c}(\phi_3) & V_{ij\theta}^{1s}(\phi_3) & \cdots & V_{ij\theta}^{Mc}(\phi_3) & V_{ij\theta}^{Ms}(\phi_3) \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ V_{ij\theta}^{0c}(\phi_{2M}) & V_{ij\theta}^{1c}(\phi_{2M}) & V_{ij\theta}^{1s}(\phi_{2M}) & \cdots & V_{ij\theta}^{Mc}(\phi_{2M}) & V_{ij\theta}^{Ms}(\phi_{2M}) \\ V_{ij\theta}^{0c}(\phi_{2M+1}) & V_{ij\theta}^{1c}(\phi_{2M+1}) & V_{ij\theta}^{1s}(\phi_{2M+1}) & \cdots & V_{ij\theta}^{Mc}(\phi_{2M+1}) & V_{ij\theta}^{Ms}(\phi_{2M+1}) \end{bmatrix} \quad (46)$$

1 where ϕ_k ($k=1, 2, 3, \dots, 2M+1$) is the k th collocation angle of the collocation points on each boundary and the elements of the sub-matrix are defined as follows:

$$U_{ij}^{nc}(\phi_k) = \int_{B_j} U(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (47)$$

$$U_{ij}^{ns}(\phi_k) = \int_{B_j} U(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (48)$$

$$\Theta_{ij}^{nc}(\phi_k) = \int_{B_j} \Theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (49)$$

$$\Theta_{ij}^{ns}(\phi_k) = \int_{B_j} \Theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (50)$$

$$M_{ij}^{nc}(\phi_k) = \int_{B_j} M(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (51)$$

$$M_{ij}^{ns}(\phi_k) = \int_{B_j} M(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (52)$$

$$V_{ij}^{nc}(\phi_k) = \int_{B_j} V(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (53)$$

$$V_{ij}^{ns}(\phi_k) = \int_{B_j} V(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (54)$$

$$U_{ij\theta}^{nc}(\phi_k) = \int_{B_j} U_\theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (55)$$

$$U_{ij\theta}^{ns}(\phi_k) = \int_{B_j} U_\theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (56)$$

$$\Theta_{ij\theta}^{nc}(\phi_k) = \int_{B_j} \Theta_\theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (57)$$

$$\Theta_{ij\theta}^{ns}(\phi_k) = \int_{B_j} \Theta_\theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (58)$$

$$M_{ij\theta}^{nc}(\phi_k) = \int_{B_j} M_\theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (59)$$

$$M_{ij\theta}^{ns}(\phi_k) = \int_{B_j} M_\theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (60)$$

$$V_{ij\theta}^{nc}(\phi_k) = \int_{B_j} V_\theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (61)$$

$$V_{ij\theta}^{ns}(\phi_k) = \int_{B_j} V_\theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (62)$$

1 where the interior degenerate kernels are used for $i = 1$ and $j = 1, 2$; the exterior degenerate kernels
 2 are used for $i = 2$ and $j = 1, 2$. However, the stream function on the boundary of inner rotating
 3 cylinder is an unknown constant u_1 [13, 14] for the viscous flow problems. In other words, one more
 4 unknown degree of freedom is introduced in the real implementation. Therefore, an extra constraint
 5 is required to uniquely solve the problem. The additional equation is obtained on the physical
 6 view that the pressure is periodic in 2π around the inner rotating cylinder. According to the Stokes
 7 equation of motion and $\nabla^2 u = \omega$, the pressure P and vorticity ω satisfy the Cauchy–Riemann
 equation, the condition for periodicity in P , namely

$$9 \quad \int_{B_1} \frac{\partial P}{\partial t} dB_1 = 0 \quad (63)$$

becomes

$$11 \quad \int_{B_1} \frac{\partial \omega}{\partial n} dB_1 = \int_{B_1} \omega_n dB_1 = 0 \quad (64)$$

12 where ω_n is the normal derivative of vorticity, t and n are tangent and normal vectors on the
 13 boundary for the Cauchy–Riemann relation. If u is solved, the vorticity can be determined by
 14 $\omega = \nabla^2 u$ in the post-processing using Equation (18). Therefore, ω_n can be obtained by taking
 15 normal derivative with respect to $\omega(x)$ in Equation (18)

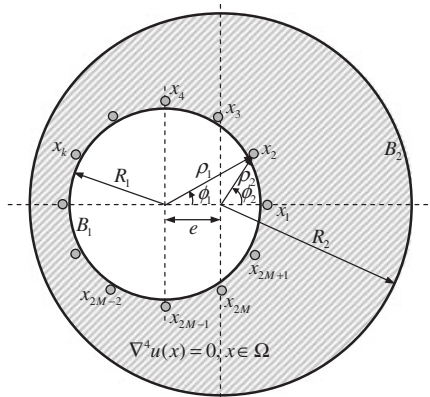
$$\omega_n = \frac{1}{8\pi} \sum_{j=1}^{N_C} \int_{B_j} \{-U_{\nabla^2, n}(s, x)v_j(s) + \Theta_{\nabla^2, n}(s, x)m_j(s) \\ - M_{\nabla^2, n}(s, x)\theta_j(s) + V_{\nabla^2, n}(s, x)u_j(s)\} dB_j(s) \quad (65)$$

16 in which $U_{\nabla^2, n}(s, x)$, $\Theta_{\nabla^2, n}(s, x)$, $M_{\nabla^2, n}(s, x)$ and $V_{\nabla^2, n}(s, x)$ are the normal derivatives of
 17 Laplacian of the degenerate kernels $U(s, x)$, $\Theta(s, x)$, $M(s, x)$ and $V(s, x)$, respectively, which are
 18 listed in Appendix A, N_C is the number of circular boundaries. By substituting Equation (65) into
 19 Equation (64), we have the constraint equation

$$\int_{B_1} \left\{ \sum_{j=1}^{N_C} \int_{B_j} [-U_{\nabla^2, n}(s, x)v_j(s) + \Theta_{\nabla^2, n}(s, x)m_j(s) \\ - M_{\nabla^2, n}(s, x)\theta_j(s) + V_{\nabla^2, n}(s, x)u_j(s)] dB_j(s) \right\} dB_1(x) = 0 \quad (66)$$

20 Equation (66) indicates that the constraint is composed of double boundary integrals. It is
 21 noted that the point x in the first boundary integral is located by approaching x from the
 22 domain to R_1^+ as shown in Figure 6. For the double integration of the same inner boundaries
 23 $\int_{B_1} \int_{B_1}$, the analytical integration can be obtained by using the orthogonal property of Fourier
 24 bases. For the double integration on different boundaries $\int_{B_1} \int_{B_2}$, trapezoid integral is used
 25 as follows:

$$\int_0^{2\pi} f(\phi) d\phi = \sum_{k=1}^N \frac{2\pi}{N} f(\phi_k) \quad (67)$$



○: Collocation point x to B_1 from R_1^+

Figure 6. Collocation method for the constraint equation.

1 where the outer boundary is uniformly divided into N segments. By matching the boundary
 2 conditions at the $2M+1$ collocation points on each boundary and rearranging the known and
 3 unknown sets, the linear algebraic system Equation (39) is reformulated to

$$\begin{bmatrix}
 \mathbf{U11} & \mathbf{\Theta11} & \mathbf{U12} & \mathbf{\Theta12} & \mathbf{V11} \\
 \mathbf{U11}_\theta & \mathbf{\Theta11}_\theta & \mathbf{U12}_\theta & \mathbf{\Theta12}_\theta & \mathbf{V11}_\theta \\
 \mathbf{U21} & \mathbf{\Theta21} & \mathbf{U22} & \mathbf{\Theta22} & \mathbf{V21} \\
 \mathbf{U21}_\theta & \mathbf{\Theta21}_\theta & \mathbf{U22}_\theta & \mathbf{\Theta22}_\theta & \mathbf{V21}_\theta \\
 \mathbf{U11}_{\nabla^2, n} & \mathbf{\Theta11}_{\nabla^2, n} & \mathbf{U12}_{\nabla^2, n} & \mathbf{\Theta12}_{\nabla^2, n} & \mathbf{V11}_{\nabla^2, n}
 \end{bmatrix}
 \begin{Bmatrix}
 \mathbf{v}_1 \\
 \mathbf{m}_1 \\
 \mathbf{v}_2 \\
 \mathbf{m}_2 \\
 u_1
 \end{Bmatrix}
 = \theta_1
 \begin{Bmatrix}
 \mathbf{M11} \\
 \mathbf{M11}_\theta \\
 \mathbf{M21} \\
 \mathbf{M21}_\theta \\
 \mathbf{M11}_{\nabla^2, n}
 \end{Bmatrix}
 \quad (68)$$

5 where $\theta_1 = \omega_1 r_1$ due to the rotation of inner cylinder [13, 14]. It is noted that $[\mathbf{V12}]$, $[\mathbf{V12}_\theta]$,
 6 $[\mathbf{V22}]$, $[\mathbf{V22}_\theta]$, $[\mathbf{V12}_{\nabla^2, n}]$, $[\mathbf{M12}]$, $[\mathbf{M12}_\theta]$, $[\mathbf{M22}]$, $[\mathbf{M22}_\theta]$ and $[\mathbf{M12}_{\nabla^2, n}]$ disappear since the
 7 outer cylinder is stationary ($u_2 = 0$ and $\theta_2 = 0$). The sub-matrices $[\mathbf{U11}_{\nabla^2, n}]$, $[\mathbf{\Theta11}_{\nabla^2, n}]$, $[\mathbf{U12}_{\nabla^2, n}]$
 and $[\mathbf{\Theta12}_{\nabla^2, n}]$ with a dimension of one by $(2M+1)$ are shown below:

$$[\mathbf{U11}_{\nabla^2, n}] = [U11_{\nabla^2, n}^{0c} \quad U11_{\nabla^2, n}^{1c} \quad U11_{\nabla^2, n}^{1s} \quad \cdots \quad U11_{\nabla^2, n}^{Mc} \quad U11_{\nabla^2, n}^{Ms}] \quad (69)$$

$$[\mathbf{\Theta11}_{\nabla^2, n}] = [\Theta11_{\nabla^2, n}^{0c} \quad \Theta11_{\nabla^2, n}^{1c} \quad \Theta11_{\nabla^2, n}^{1s} \quad \cdots \quad \Theta11_{\nabla^2, n}^{Mc} \quad \Theta11_{\nabla^2, n}^{Ms}] \quad (70)$$

$$[\mathbf{U12}_{\nabla^2, n}] = [U12_{\nabla^2, n}^{0c} \quad U12_{\nabla^2, n}^{1c} \quad U12_{\nabla^2, n}^{1s} \quad \cdots \quad U12_{\nabla^2, n}^{Mc} \quad U12_{\nabla^2, n}^{Ms}] \quad (71)$$

$$[\mathbf{\Theta12}_{\nabla^2, n}] = [\Theta12_{\nabla^2, n}^{0c} \quad \Theta12_{\nabla^2, n}^{1c} \quad \Theta12_{\nabla^2, n}^{1s} \quad \cdots \quad \Theta12_{\nabla^2, n}^{Mc} \quad \Theta12_{\nabla^2, n}^{Ms}] \quad (72)$$

1 where each element of $[U11_{\nabla^2,n}]$, $[\Theta11_{\nabla^2,n}]$, $[U12_{\nabla^2,n}]$ and $[\Theta12_{\nabla^2,n}]$ are defined as shown below:

$$U11_{\nabla^2,n}^{nc} = \int_{B_1} \int_{B_1} U_{\nabla^2,n}^E(s, x) \cos(n\theta_1) dB_1(s) dB_1(x), \quad n=0, 1, 2, 3, \dots, M \quad (73)$$

$$U11_{\nabla^2,n}^{ns} = \int_{B_1} \int_{B_1} U_{\nabla^2,n}^E(s, x) \sin(n\theta_1) dB_1(s) dB_1(x), \quad n=1, 2, 3, \dots, M \quad (74)$$

$$\Theta11_{\nabla^2,n}^{nc} = \int_{B_1} \int_{B_1} \Theta_{\nabla^2,n}^E(s, x) \cos(n\theta_1) dB_1(s) dB_1(x), \quad n=0, 1, 2, 3, \dots, M \quad (75)$$

$$\Theta11_{\nabla^2,n}^{ns} = \int_{B_1} \int_{B_1} \Theta_{\nabla^2,n}^E(s, x) \sin(n\theta_1) dB_1(s) dB_1(x), \quad n=1, 2, 3, \dots, M \quad (76)$$

$$\begin{aligned} U12_{\nabla^2,n}^{nc} &= \int_{B_1} \int_{B_2} U_{\nabla^2,n}^I(s, x_k) \cos(n\theta_2) dB_2(s) dB_1(x) \\ &= \sum_{k=1}^N \frac{2\pi}{N} \int_{B_2} U_{\nabla^2,n}^I(s, x_k) \cos(n\theta_2) dB_2(s), \quad n=0, 1, 2, 3, \dots, M \end{aligned} \quad (77)$$

$$\begin{aligned} U12_{\nabla^2,n}^{ns} &= \int_{B_1} \int_{B_2} U_{\nabla^2,n}^I(s, x_k) \sin(n\theta_2) dB_2(s) dB_1(x) \\ &= \sum_{k=1}^N \frac{2\pi}{N} \int_{B_2} U_{\nabla^2,n}^I(s, x_k) \sin(n\theta_2) dB_2(s), \quad n=1, 2, 3, \dots, M \end{aligned} \quad (78)$$

$$\begin{aligned} \Theta12_{\nabla^2,n}^{nc} &= \int_{B_1} \int_{B_2} \Theta_{\nabla^2,n}^I(s, x_k) \cos(n\theta_2) dB_2(s) dB_1(x) \\ &= \sum_{k=1}^N \frac{2\pi}{N} \int_{B_2} \Theta_{\nabla^2,n}^I(s, x_k) \cos(n\theta_2) dB_2(s), \quad n=0, 1, 2, 3, \dots, M \end{aligned} \quad (79)$$

$$\begin{aligned} \Theta12_{\nabla^2,n}^{ns} &= \int_{B_1} \int_{B_2} \Theta_{\nabla^2,n}^I(s, x_k) \sin(n\theta_2) dB_2(s) dB_1(x) \\ &= \sum_{k=1}^N \frac{2\pi}{N} \int_{B_2} \Theta_{\nabla^2,n}^I(s, x_k) \sin(n\theta_2) dB_2(s), \quad n=1, 2, 3, \dots, M \end{aligned} \quad (80)$$

3

1 where x_k is sampling point. The elements of $[M11_{\nabla^2, n}]$ and $[V11_{\nabla^2, n}]$ with a dimension of one
 by one are defined as follows:

$$M11_{\nabla^2, n} = \int_{B_1} \int_{B_1} M_{\nabla^2, n}^E(s, x) 1 dB(s)_1 dB_1(x) \quad (81)$$

$$V11_{\nabla^2, n} = \int_{B_1} \int_{B_1} V_{\nabla^2, n}^E(s, x) 1 dB_1(s) dB_1(x) \quad (82)$$

3 The unknown Fourier coefficients and the unknown stream function on the inner rotating cylinder
 5 can be obtained at the same time by solving the linear algebraic augmented system of Equation
 (68). After determining the unknown Fourier coefficients, the interior potential can be obtained by
 7 using the BIE for the domain point. The vorticity in the post-processing can be obtained by using
 the following equation:

$$\begin{aligned} \nabla^2 u(x) = \omega(x) = \frac{1}{8\pi} \sum_{j=1}^{N_C} \left\{ - \int_{B_j} U_{\nabla^2}(s, x) v_j(s) dB_j(s) + \int_{B_j} \Theta_{\nabla^2}(s, x) v_j(s) dB_j(s) \right. \\ \left. - \int_{B_j} M_{\nabla^2}(s, x) v_j(s) dB_j(s) + \int_{B_j} V_{\nabla^2}(s, x) v_j(s) dB_j(s) \right\}, \quad x \in \Omega \end{aligned} \quad (83)$$

where N_C is the number of circular boundaries.

9 *6.2. Indirect formulation*

By using the indirect formulation of Equations (28)–(29) and collocating to the boundaries from
 11 R^+ and R^- for the inner and outer boundaries, respectively, the linear algebraic system is obtained
 as follows:

$$\begin{bmatrix} U11 & \Theta11 & U12 & \Theta12 \\ U11_\theta & \Theta11_\theta & U12_\theta & \Theta12_\theta \\ U21 & \Theta21 & U22 & \Theta22 \\ U21_\theta & \Theta21_\theta & U22_\theta & \Theta22_\theta \end{bmatrix} \begin{bmatrix} \Phi_1 \\ \Psi_1 \\ \Phi_2 \\ \Psi_2 \end{bmatrix} = \begin{bmatrix} u_1 \\ \theta_1 \\ u_2 \\ \theta_2 \end{bmatrix} \quad (84)$$

13 where Φ_1, Ψ_1, Φ_2 and Ψ_2 are the column vectors of Fourier coefficients for the fictitious boundary
 15 distributions of Φ and Ψ ; u_1, θ_1, u_2 and θ_2 are the given boundary conditions. The sub-matrices
 17 $[Uij], [\Thetaij], [Uij_\theta]$ and $[\Thetaij_\theta]$ ($i=1, 2$ and $j=1, 2$) of the influence matrix are the same as
 Equations (40) and (44)–(45). The elements of the sub-matrices are defined as follows:

$$Uij^{nc}(\phi_k) = \int_{B_j} U(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (85)$$

$$Uij^{ns}(\phi_k) = \int_{B_j} U(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (86)$$

$$\Theta i j^{nc}(\phi_k) = \int_{B_j} \Theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (87)$$

$$\Theta i j^{ns}(\phi_k) = \int_{B_j} \Theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (88)$$

$$U i j_0^{nc}(\phi_k) = \int_{B_j} U_\theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (89)$$

$$U i j_0^{ns}(\phi_k) = \int_{B_j} U_\theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (90)$$

$$\Theta i j_0^{nc}(\phi_k) = \int_{B_j} \Theta_\theta(s, x_k) \cos(n\theta_j) R_j d\theta_j, \quad n=0, 1, 2, 3, \dots, M \quad (91)$$

$$\Theta i j_0^{ns}(\phi_k) = \int_{B_j} \Theta_\theta(s, x_k) \sin(n\theta_j) R_j d\theta_j, \quad n=1, 2, 3, \dots, M \quad (92)$$

1 where $j=1$ and $i=1, 2$, the exterior degenerate kernels are used; $j=2$ and $i=1, 2$, the interior
 3 degenerate kernels are used. However, u_1 is an unknown constant along the inner cylinder as
 explained in the direct BIEM, one more constraint equation is needed and Equation (64) is
 considered again as follows:

$$5 \int_{B_1} \frac{\partial \omega}{\partial n} dB_1 = \int_{B_1} \omega_n dB_1 = 0 \quad (93)$$

By substituting Equation (32) into Equation (93), we have

$$7 \int_{B_1} \sum_{j=1}^{N_C} \left\{ \int_{B_j} U_{\nabla^2, n}(s, x) \Phi_j(s) dB_j(s) + \int_{B_j} \Theta_{\nabla^2, n}(s, x) \Psi_j(s) dB_j(s) \right\} dB_1(x) = 0 \quad (94)$$

Therefore, the linear algebraic system (84) can be reformulated as shown below:

$$9 \begin{bmatrix} U11 & \Theta11 & U12 & \Theta12 & -1 \\ U11_0 & \Theta11_0 & U12_0 & \Theta12_0 & 0 \\ U21 & \Theta21 & U22 & \Theta22 & 0 \\ U21_0 & \Theta21_0 & U22_0 & \Theta22_0 & 0 \\ U11_{\nabla^2, n} & \Theta11_{\nabla^2, n} & U12_{\nabla^2, n} & \Theta12_{\nabla^2, n} & 0 \end{bmatrix} \begin{Bmatrix} \Phi_1 \\ \Psi_1 \\ \Phi_2 \\ \Psi_2 \\ u_1 \end{Bmatrix} = \begin{Bmatrix} 0 \\ \theta_1 \\ u_2 \\ \theta_2 \\ 0 \end{Bmatrix} \quad (95)$$

11 The sub-matrices $[U11_{\nabla^2, n}]$, $[\Theta11_{\nabla^2, n}]$, $[U12_{\nabla^2, n}]$, $[\Theta12_{\nabla^2, n}]$ with a dimension of one by $(2M + 1)$,
 12 respectively, are the same as Equations (69)–(72). The unknown Fourier coefficients and the
 13 unknown stream function along the inner rotating cylinder can be obtained at the same time by
 solving the linear algebraic augmented system of Equation (95). After determining the unknown
 Fourier coefficients, the interior potential can be obtained by using the BIE for the domain point

1 of Equation (28). The vorticity in the post-processing can be obtained by using the following
 2 equation:

$$3 \quad \omega(x) = \sum_{j=1}^{N_C} \left\{ \int_{B_j} U_{\nabla^2}(s, x) \Phi_j(s) dB_j(s) + \int_{B_j} \Theta_{\nabla^2}(s, x) \Psi_j(s) dB_j(s) \right\}, \quad x \in \Omega \quad (96)$$

where N_C is the number of circular boundaries.

5 **7. NUMERICAL EXAMPLES**

7.1. *Eccentric case: a doubly connected domain*

7 Two approaches, direct BIEM and indirect BIEM, are presented to solve the flow between eccen-
 8 tric cylinders. The inner cylinder rotates with a constant angular velocity and the outer one is
 9 stationary as shown in Figure 7. The following parameters are defined: $r_1=0.5$, radius of inner
 10 cylinder; $r_2=1$, radius of outer cylinder; $c=r_2-r_1$, the clearance; $\varepsilon=e/c$, the eccentricity; e ,
 11 separation of centers of cylinders; $\omega_1=1$ for the anticlockwise angular velocity of inner cylinder.

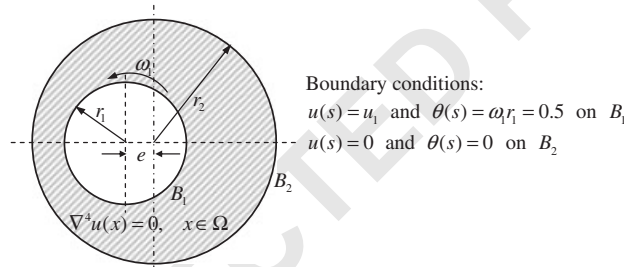


Figure 7. The flow between eccentric cylinders.

Table I. Comparison of analytical and numerical results of u_1 for the eccentric bearing.

Kelmanson and Ingham [13, 14]					Analytical solution	Present method (Direct BIEM)	Present method (Indirect BIEM)
ε	$n = 80$	$n = 160$	$n = 320$	Limit $n \rightarrow \infty$ [1]			
0.0	0.1066	0.1062	0.1061	0.1061	0.1060	0.1060 ($N=5$)	0.1060 ($N=5$)
0.1	0.1052	0.1048	0.1047	0.1047	0.1046	0.1046 ($N=7$)	0.1046 ($N=7$)
0.2	0.1011	0.1006	0.1005	0.1005	0.1005	0.1005 ($N=7$)	0.1005 ($N=7$)
0.3	0.0944	0.0939	0.0938	0.0938	0.0938	0.0938 ($N=7$)	0.0938 ($N=7$)
0.4	0.0854	0.0850	0.0848	0.0846	0.0848	0.0848 ($N=9$)	0.0848 ($N=9$)
0.5	0.0748	0.0740	0.0739	0.0739	0.0738	0.0738 ($N=11$)	0.0738 ($N=11$)
0.6	0.0622	0.0615	0.0613	0.0612	0.0611	0.0611 ($N=17$)	0.0611 ($N=17$)
0.7	0.0484	0.0477	0.0474	0.0472	0.0472	0.0472 ($N=17$)	0.0472 ($N=17$)
0.8	0.0347	0.0332	0.0326	0.0322	0.0322	0.0322 ($N=21$)	0.0322 ($N=21$)
0.9	0.0191	0.0175	0.0168	0.0163	0.0164	0.0164 ($N=31$)	0.0164 ($N=31$)

n , the number of boundary nodes; N , the number of collocation points on the inner cylinder.

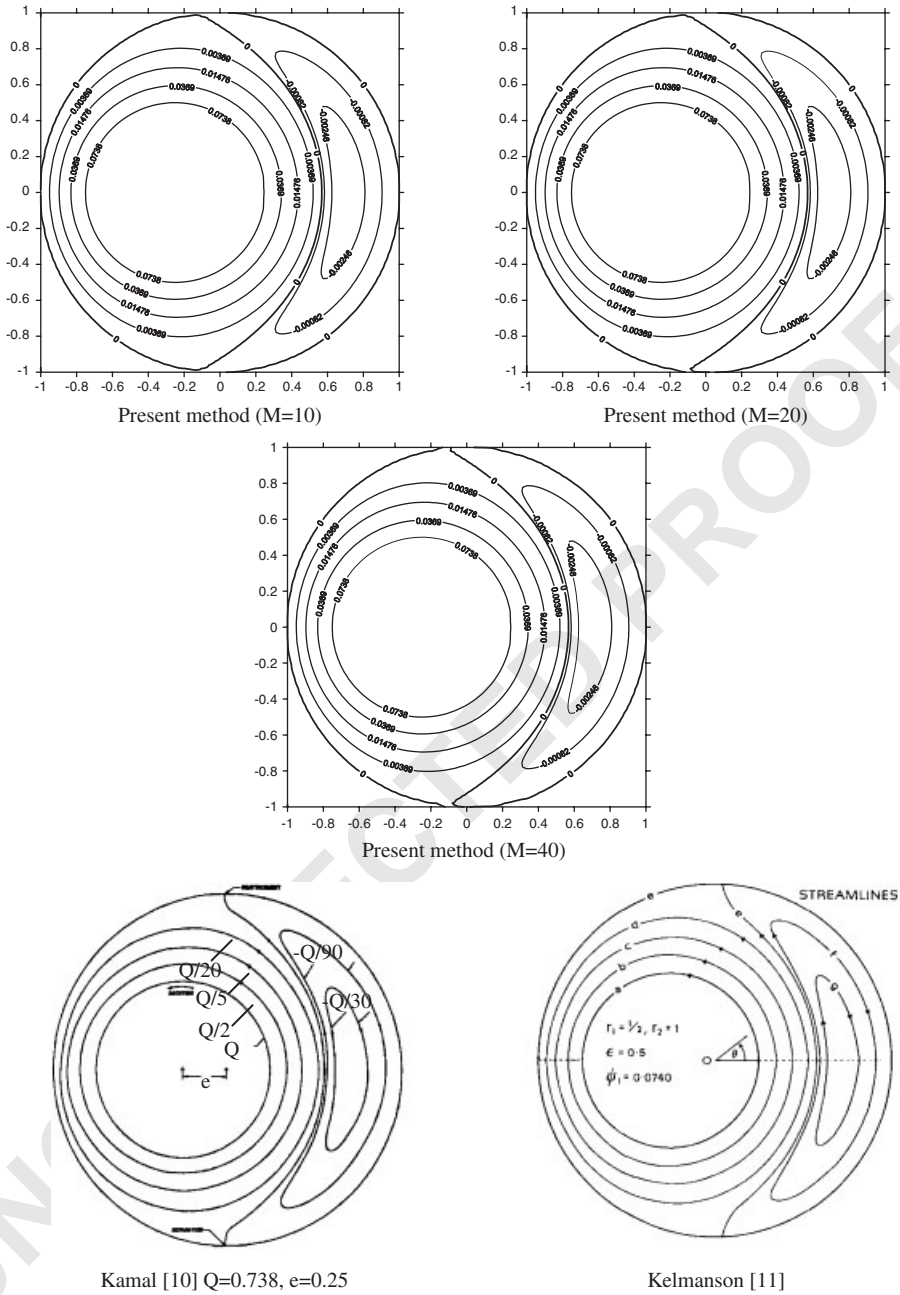


Figure 8. Comparison of contour plots of streamlines for $\epsilon=0.5$.

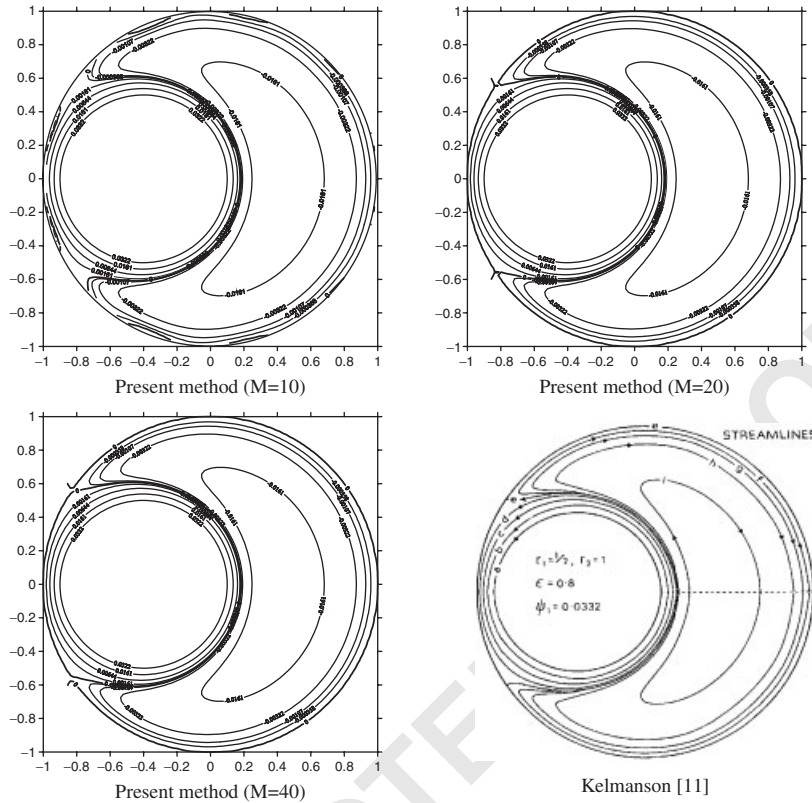


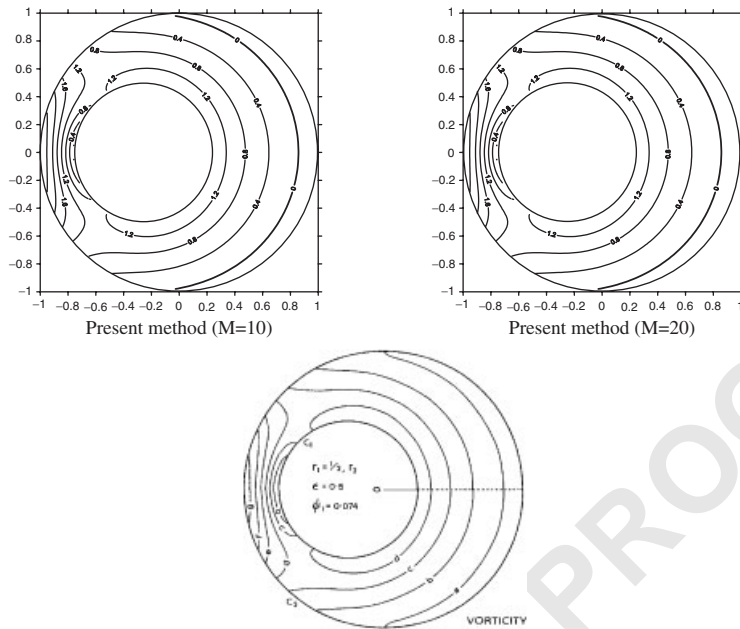
Figure 9. Comparison of streamlines contour plots for $\varepsilon=0.8$.

- 1 The flow between eccentric cylinders satisfies the biharmonic equation and the essential boundary conditions are specified as follows:

$$u(s) = u_1, \quad \theta(s) = \frac{\partial u(s)}{\partial n} = \omega_1 r_1 = 0.5, \quad s \text{ on } B_1 \tag{97}$$

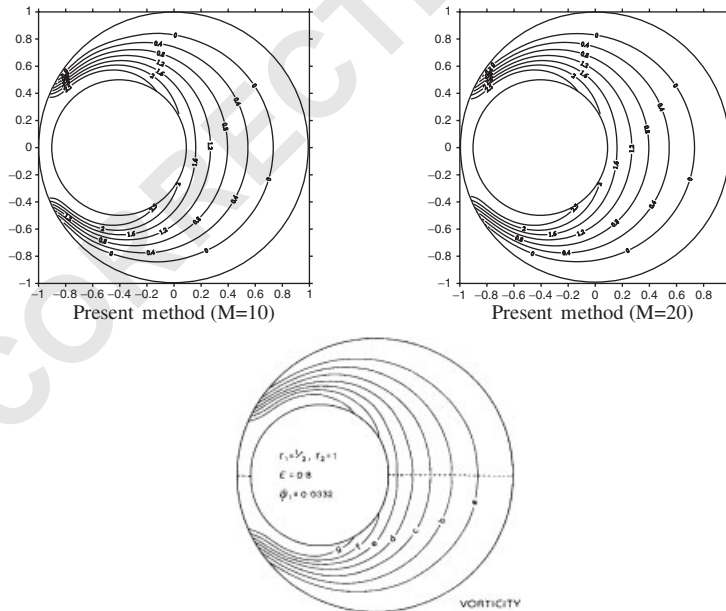
$$u(s) = 0, \quad \theta(s) = \frac{\partial u(s)}{\partial n} = 0, \quad s \text{ on } B_2 \tag{98}$$

- 3 First, the direct BIEM is used. The unknown boundary densities $m(s)$, $v(s)$ on B_1 and $m(s)$, $v(s)$ on B_2 are expressed in terms of Fourier series. The unknown Fourier coefficients can be determined
- 5 by using the null-field integral equations in conjunction with degenerate kernels and Fourier series; however, the boundary condition u_1 is an unknown constant along the inner boundary. An additional
- 7 constraint is required to ensure a unique solution. From the solution procedures of the direct BIEM, u_1 with different eccentricities are calculated and the results are shown in Table I. By using the



Kelmanson [11] (a) 0, (b) 0.4, (c) 0.8, (d) 1.2, (e) 1.6, (f) 2.0, (g) 2.5

Figure 10. Comparison of vorticity contour plots for $\epsilon=0.5$.



Kelmanson [11] (a) 0, (b) 0.4, (c) 0.8, (d) 1.2, (e) 1.6, (f) 2.0, (g) 2.5

Figure 11. Comparison of vorticity contour plots for $\epsilon=0.8$.

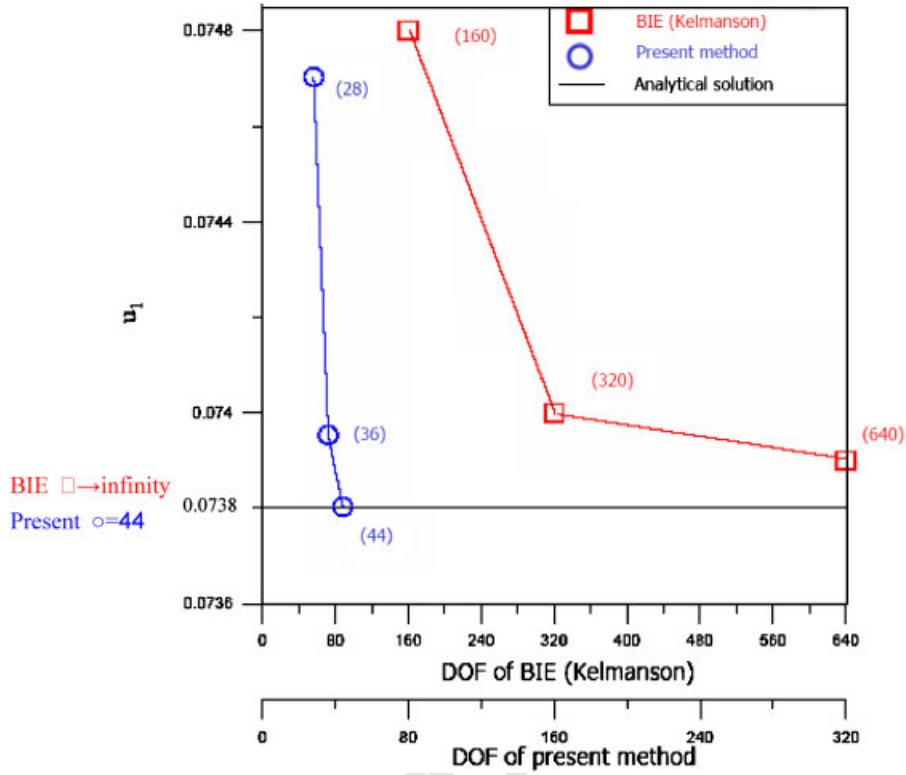


Figure 12. Comparison for $\varepsilon=0.5$ using direct formulation.

- 1 fewer degrees of freedom than BIE [14], present results are more accurate after comparing with the analytical solution as follows:

$$u_1 = \frac{A \omega_1 r_1 (\sinh \delta - \delta \cosh \delta) (\sinh \alpha_2 \sinh \delta - \delta \sinh \alpha_1)}{2[(\delta + \sinh \alpha_1 \cosh \alpha_1 - \cosh \alpha_2 \sinh \alpha_2) (\sinh \delta - \delta \cosh \delta) + \cosh \delta (\delta^2 - \sinh^2 \delta)]} \quad (99)$$

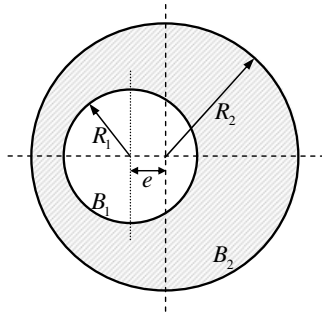
where

$$A = \frac{c}{\varepsilon} \left[(1 - \varepsilon^2) \left[\left(\frac{r_1 + r_2}{c} \right)^2 - \varepsilon^2 \right] \right]^{1/2} \quad (100)$$

$$\alpha_1 = -\sinh^{-1} \left(\frac{A}{2r_1} \right) \quad (101)$$

$$\alpha_2 = -\sinh^{-1} \left(\frac{A}{2r_2} \right) \quad (102)$$

$$\delta = \alpha_1 - \alpha_2 \quad (103)$$



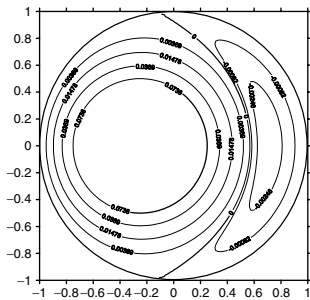
Geometric data:

$R_1 = 0.5, R_2 = 1, e = 0.25$

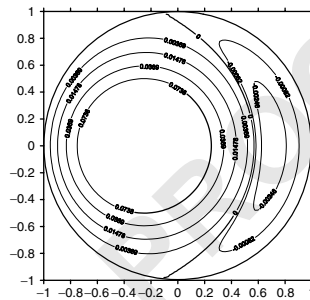
Boundary conditions:

$u_1(s) = 0.0738$ and $\theta_1(s) = 0.5$ on B_1

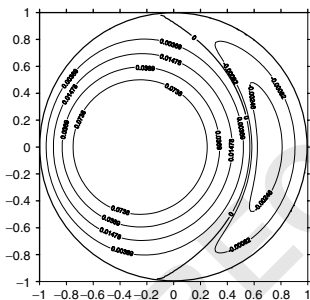
$u_2(s) = 0$ and $\theta_2(s) = 0$ on B_2



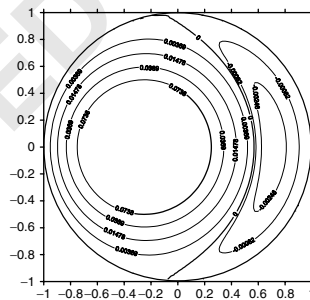
M=10



M=20



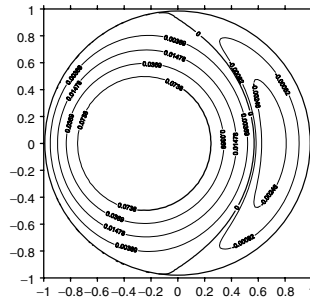
M=40



M=80

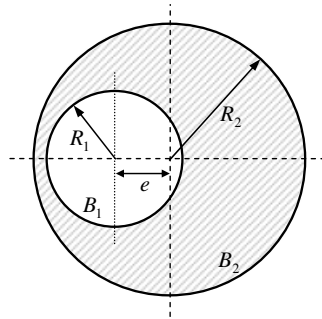


FEM mesh (No. of nodes=1,734
No. of elements=3,218)



FEM result

Figure 13. The streamlines contour plot for $\varepsilon = 0.5$ by using indirect BIEM.



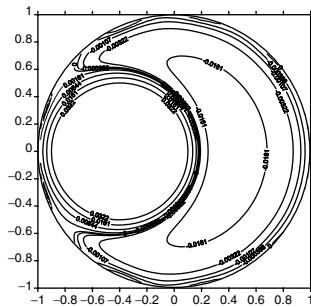
Geometric data:

$R_1 = 0.5, R_2 = 1, e = 0.4$

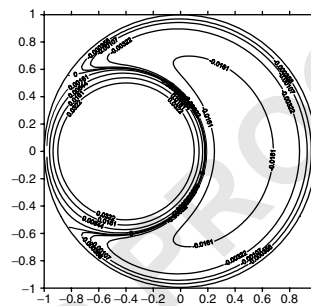
Boundary conditions:

$u_1(s) = 0.0322$ and $\theta_1(s) = 0.5$ on B_1

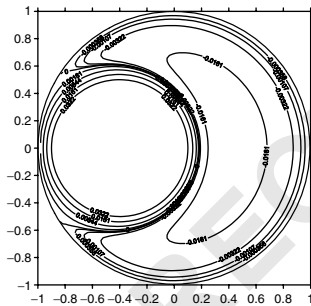
$u_2(s) = 0$ and $\theta_2(s) = 0$ on B_2



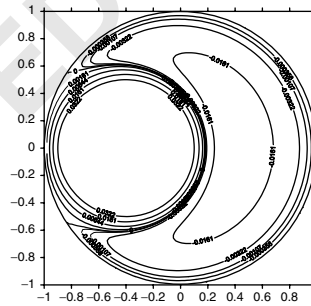
M=10



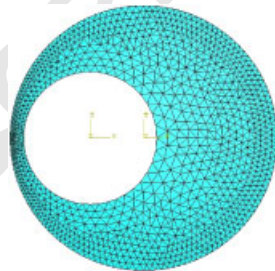
M=20



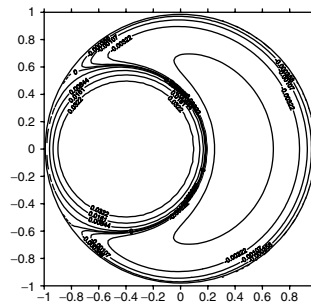
M=40



M=80



FEM mesh (No. of nodes=1,196
No. of elements=2,162)



FEM result

Figure 14. The streamlines contour plot for $\varepsilon = 0.8$ by using indirect BIEM.

1 It is noted that the number of segments N in integrating on B_1 boundary of Equation (68) is the
 3 same as the number of $(2M + 1)$ collocation null-field points near the inner cylinder boundary. The
 contour plot of stream function and vorticity can be obtained by substituting Fourier coefficients
 5 into the BIE for the domain point of Equations (9) and (18). The streamlines and vorticity contour
 plots for $\varepsilon = 0.5$ and 0.8 solved by employing the direct BIEM are compared with the Kelmanson's
 7 results [14] obtained by using the 160 boundary nodes and Kamal's result [11] as shown in Figures
 8–11. Figure 12 shows the rate of convergence between the present approach and BIE. It indicates
 that our approach shows exponential convergence rate.

9 According to the indirect BIEM, the unknown boundary constant u_1 for the eccentric bearing
 problem is also obtained as shown in Table I. Good agreement is also made after comparing the
 11 results for $\varepsilon = 0.5$ and 0.8 as shown in Figures 13 and 14. Besides, the FEM by using ABAQUS
 software is used to solve the problem and the results are also shown in Figures 13 and 14 for
 13 comparison.

8. CONCLUDING REMARKS

15 In this paper, the direct and indirect formulations in conjunction with the degenerate kernels and
 Fourier series expansion in adaptive observer system were proposed to solve the Stokes flow
 17 problems. ABAQUS software [18] was also used to solve the stream function for the eccentric
 bearing case. The constant stream function along the inner rotating cylinder is obtained by using
 19 direct and indirect BIEMs. Only fewer numbers of collocation and segments were used to show the
 good agreement after comparing with the BIE results [13, 14] on the base of analytical solution.
 21 Although the Poisson ratio is contained in the direct BIEM, this method can be applied to solve
 the Stokes problems no matter how the Poisson ratio is specified. Although the indirect BIEM
 23 cannot provide null-field integral equation, the present method by moving the interior point to
 the boundary can be implemented by choosing the appropriate expansion of degenerate kernels.
 25 Numerical examples were demonstrated to see the validity of the present formulation with five
 gains: meshless approach, boundary-layer effect free, singularity free, exponential convergence
 27 and well-posed model.

APPENDIX A: DEGENERATE KERNELS

29 *A.1. Degenerate kernels for U, Θ, M, V in the first BIE*

$$U(s, x) = \begin{cases} U^I(s, x) = \rho^2(1 + \ln R) + R^2 \ln R - \left[R\rho(1 + 2 \ln R) + \frac{1}{2} \frac{\rho^3}{R} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{1}{m(m+1)} \frac{\rho^{m+2}}{R^m} - \frac{1}{m(m-1)} \frac{\rho^m}{R^{m-2}} \right] \cos[m(\theta - \phi)], & R \geq \rho \\ U^E(s, x) = R^2(1 + \ln \rho) + \rho^2 \ln \rho - \left[\rho R(1 + 2 \ln \rho) + \frac{1}{2} \frac{R^3}{\rho} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{1}{m(m+1)} \frac{R^{m+2}}{\rho^m} - \frac{1}{m(m-1)} \frac{R^m}{\rho^{m-2}} \right] \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$\Theta(s, x) = \begin{cases} \Theta^I(s, x) = \frac{\rho^2}{R} + R(1 + 2\ln R) - \left[\rho(3 + 2\ln R) - \frac{1}{2} \frac{\rho^3}{R^2} \right] \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[\frac{1}{m+1} \frac{\rho^{m+2}}{R^{m+1}} - \frac{m-2}{m(m-1)} \frac{\rho^m}{R^{m-1}} \right] \cos[m(\theta - \phi)], \quad R \geq \rho \\ \Theta^E(s, x) = 2R(1 + \ln \rho) - \left[\rho(1 + 2\ln \rho) + \frac{3}{2} \frac{R^2}{\rho} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{m+2}{m(m+1)} \frac{R^{m+1}}{\rho^m} - \frac{1}{m-1} \frac{R^{m-1}}{\rho^{m-2}} \right] \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

$$M(s, x) = \begin{cases} M^I(s, x) = (v-1) \frac{\rho^2}{R^2} + (v+3) + 2(v+1) \ln R - \left[(v+1) \frac{2\rho}{R} - (v-1) \frac{\rho^3}{R^3} \right] \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[(v-1) \frac{\rho^{m+2}}{R^{m+2}} + \frac{m(1-v) - 2(1+v)}{m} \frac{\rho^m}{R^m} \right] \cos[m(\theta - \phi)], \quad R \geq \rho \\ M^E(s, x) = 2(1+v)(1 + \ln \rho) - (v+3) \frac{R}{\rho} \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[\frac{m(v-1) - 2(v+1)}{m} \frac{R^m}{\rho^m} + (1-v) \frac{R^{m-2}}{\rho^{m-2}} \right] \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

$$V(s, x) = \begin{cases} V^I(s, x) = \frac{4}{R} + \left[\frac{2\rho}{R^2} (3-v) - \frac{\rho^3}{R^4} (1-v) \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[m(1-v) \frac{\rho^{m+2}}{R^{m+3}} - (4+m(1-v)) \frac{\rho^m}{R^{m+1}} \right] \cos[m(\theta - \phi)], \quad R > \rho \\ V^E(s, x) = (-3-v) \frac{1}{\rho} \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[(m(1-v) - 4) \frac{R^{m-1}}{\rho^m} - m(1-v) \frac{R^{m-3}}{\rho^{m-2}} \right] \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

1

A.2. Degenerate kernels for U_θ , Θ_θ , V_θ in the second BIE

$$U_\theta(s, x) = \begin{cases} U_\theta^I(s, x) = 2\rho(1 + \ln R) - \left[R(1 + 2\ln R) + \frac{3}{2} \frac{\rho^2}{R} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{m+2}{m(m+1)} \frac{\rho^{m+1}}{R^m} - \frac{1}{m-1} \frac{\rho^{m-1}}{R^{m-2}} \right] \cos[m(\theta - \phi)], \quad R \geq \rho \\ U_\theta^E(s, x) = \frac{R^2}{\rho} + \rho(1 + 2\ln \rho) - \left[R(3 + 2\ln \rho) - \frac{1}{2} \frac{R^3}{\rho^2} \right] \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[\frac{1}{m+1} \frac{R^{m+2}}{\rho^{m+1}} - \frac{m-2}{m(m-1)} \frac{R^m}{\rho^{m-1}} \right] \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

$$\Theta_{\theta}(s, x) = \begin{cases} \Theta_{\theta}^I(s, x) = \frac{2\rho}{R} - \left[(3+2\ln R) - \frac{3}{2} \frac{\rho^2}{R^2} \right] \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[\frac{m+2}{m+1} \frac{\rho^{m+1}}{R^{m+1}} - \frac{m-2}{m-1} \frac{\rho^{m-1}}{R^{m-1}} \right] \cos[m(\theta - \phi)], \quad R \geq \rho \\ \Theta_{\theta}^E(s, x) = \frac{2R}{\rho} - \left[(3+2\ln \rho) - \frac{3}{2} \frac{R^2}{\rho^2} \right] \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[\frac{m+2}{m+1} \frac{R^{m+1}}{\rho^{m+1}} - \frac{m-2}{m-1} \frac{R^{m-1}}{\rho^{m-1}} \right] \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

$$M_{\theta}(s, x) = \begin{cases} M_{\theta}^I(s, x) = \frac{2\rho}{R^2}(v-1) - \left[\frac{2}{R}(v+1) - 3(v-1) \frac{\rho^2}{R^3} \right] \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[(m+2)(v-1) \frac{\rho^{m+1}}{R^{m+2}} + (m(1-v) - 2(1+v)) \frac{\rho^{m-1}}{R^m} \right] \\ \quad \times \cos[m(\theta - \phi)], \quad R > \rho \\ M_{\theta}^E(s, x) = \frac{2(1+v)}{\rho} + (v+3) \frac{R}{\rho^2} \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[(m(v-1) - 2(v+1)) \frac{R^m}{\rho^{m+1}} + (m-2)(1-v) \frac{R^{m-2}}{\rho^{m-1}} \right] \\ \quad \times \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

$$V_{\theta}(s, x) = \begin{cases} V_{\theta}^I(s, x) = \left[\frac{2}{R^2}(3-v) - 3(1-v) \frac{\rho^2}{R^4} \right] \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[m(m+2)(1-v) \frac{\rho^{m+1}}{R^{m+3}} - m(4+m(1-v)) \frac{\rho^{m-1}}{R^{m+1}} \right] \\ \quad \times \cos[m(\theta - \phi)], \quad R > \rho \\ V_{\theta}^E(s, x) = (3+v) \frac{1}{\rho^2} \cos(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[m(m(1-v) - 4) \frac{R^{m-1}}{\rho^{m+1}} - m(m-2)(1-v) \frac{R^{m-3}}{\rho^{m-1}} \right] \\ \quad \times \cos[m(\theta - \phi)], \quad \rho > R \end{cases}$$

- 1 where U_{θ} , Θ_{θ} , M_{θ} , V_{θ} are equal to $\partial U(s, x)/\partial n_x$, $\partial \Theta(s, x)/\partial n_x$, $\partial M(s, x)/\partial n_x$ and $\partial V(s, x)/\partial n_x$, respectively.

1 *A.3. Tangential derivative with respect to the field point*

$$\begin{aligned}
 U_{,t}(s, x) &= \begin{cases} U_{,t}^I(s, x) = - \left[R(1 + 2 \ln R) + \frac{1}{2} \frac{\rho^2}{R} \right] \sin(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{1}{m+1} \frac{\rho^{m+1}}{R^m} - \frac{1}{m-1} \frac{\rho^{m-1}}{R^{m-2}} \right] \sin[m(\theta - \phi)], & R > \rho \\ U_{,t}^E(s, x) = - \left[R(1 + 2 \ln \rho) + \frac{1}{2} \frac{R^3}{\rho^2} \right] \sin(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{1}{m+1} \frac{R^{m+2}}{\rho^{m+1}} - \frac{1}{m-1} \frac{R^m}{\rho^{m-1}} \right] \sin[m(\theta - \phi)], & \rho > R \end{cases} \\
 \Theta_{,t}(s, x) &= \begin{cases} \Theta_{,t}^I(s, x) = - \left(3 + 2 \ln R - \frac{1}{2} \frac{\rho^2}{R^2} \right) \sin(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[\frac{m}{m+1} \frac{\rho^{m+1}}{R^{m+1}} - \frac{m-2}{m-1} \frac{\rho^{m-1}}{R^{m-1}} \right] \sin[m(\theta - \phi)], & R > \rho \\ \Theta_{,t}^E(s, x) = - \left(1 + 2 \ln \rho + \frac{3}{2} \frac{R^2}{\rho^2} \right) \sin(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[\frac{m+2}{m+1} \frac{R^{m+1}}{\rho^{m+1}} - \frac{m}{m-1} \frac{R^{m-1}}{\rho^{m-1}} \right] \sin[m(\theta - \phi)], & \rho > R \end{cases} \\
 M_{,t}(s, x) &= \begin{cases} M_{,t}^I(s, x) = - \left[\frac{2(v+1)}{R} - (v-1) \frac{\rho^2}{R^3} \right] \sin(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[m(v-1) \frac{\rho^{m+1}}{R^{m+2}} + (m(1-v) - 2(1+v)) \frac{\rho^{m-1}}{R^m} \right] \sin[m(\theta - \phi)], & R > \rho \\ M_{,t}^E(s, x) = - (v+3) \frac{R}{\rho^2} \sin(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} \left[(m(v-1) - 2(v+1)) \frac{R^m}{\rho^{m+1}} + m(1-v) \frac{R^{m-2}}{\rho^{m-1}} \right] \sin[m(\theta - \phi)], & \rho > R \end{cases} \\
 V_{,t}(s, x) &= \begin{cases} V_{,t}^I(s, x) = \left[\frac{2(3-v)}{R^2} - \frac{\rho^2}{R^4} (1-v) \right] \sin(\theta - \phi) \\ \quad - \sum_{m=2}^{\infty} \left[m^2(1-v) \frac{\rho^{m+1}}{R^{m+3}} - m(4+m(1-v)) \frac{\rho^{m-1}}{R^{m+1}} \right] \sin[m(\theta - \phi)], & R > \rho \\ V_{,t}^E(s, x) = (-3-v) \frac{1}{\rho^2} \sin(\theta - \phi) + \sum_{m=2}^{\infty} \left[m(m(1-v) - 4) \frac{R^{m-1}}{\rho^{m+1}} \right. \\ \quad \left. - m^2(1-v) \frac{R^{m-3}}{\rho^{m-1}} \right] \sin[m(\theta - \phi)], & \rho > R \end{cases}
 \end{aligned}$$

1 *A.4. Laplacian of the degenerate kernels with respect to U, Θ, M, V*

$$U_{\nabla^2}(s, x) = \begin{cases} U_{\nabla^2}^I(s, x) = 4(1 + \ln R) - 4\frac{\rho}{R} \cos(\theta - \phi) - \sum_{m=2}^{\infty} \frac{4}{m} \frac{\rho^m}{R^m} \cos[m(\theta - \phi)], & R > \rho \\ U_{\nabla^2}^E(s, x) = 4(1 + \ln \rho) - 4\frac{R}{\rho} \cos(\theta - \phi) - \sum_{m=2}^{\infty} \frac{4}{m} \frac{R^m}{\rho^m} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$\Theta_{\nabla^2}(s, x) = \begin{cases} \Theta_{\nabla^2}^I(s, x) = \frac{4}{R} + 4\frac{\rho}{R^2} \cos(\theta - \phi) + \sum_{m=2}^{\infty} 4\frac{\rho^m}{R^{m+1}} \cos[m(\theta - \phi)], & R > \rho \\ \Theta_{\nabla^2}^E(s, x) = -\frac{4}{\rho} \cos(\theta - \phi) - \sum_{m=2}^{\infty} 4\frac{R^{m-1}}{\rho^m} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$M_{\nabla^2}(s, x) = \begin{cases} M_{\nabla^2}^I(s, x) = \frac{4}{R^2}(v-1) + 8(v-1)\frac{\rho}{R^3} \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} 4(m+1)(v-1)\frac{\rho^m}{R^{m+2}} \cos[m(\theta - \phi)], & R > \rho \\ M_{\nabla^2}^E(s, x) = \sum_{m=2}^{\infty} 4(m-1)(v-1)\frac{R^{m-2}}{\rho^m} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$V_{\nabla^2}(s, x) = \begin{cases} V_{\nabla^2}^I(s, x) = 8(v-1)\frac{\rho}{R^4} \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} 4m(m+1)(v-1)\frac{\rho^m}{R^{m+3}} \cos[m(\theta - \phi)], & R > \rho \\ V_{\nabla^2}^E(s, x) = -\sum_{m=2}^{\infty} 4m(m-1)(v-1)\frac{R^{m-3}}{\rho^m} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

3

A.5. Normal derivative of Laplacian of the degenerate kernels

$$U_{\nabla^2, n}(s, x) = \begin{cases} U_{\nabla^2, n}^I(s, x) = -\frac{4}{R} \cos(\theta - \phi) - \sum_{m=2}^{\infty} 4\frac{\rho^{m-1}}{R^m} \cos[m(\theta - \phi)], & R > \rho \\ U_{\nabla^2, n}^E(s, x) = \frac{4}{\rho} + 4\frac{R}{\rho^2} \cos(\theta - \phi) + \sum_{m=2}^{\infty} 4\frac{R^m}{\rho^{m+1}} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$\Theta_{\nabla^2, n}(s, x) = \begin{cases} \Theta_{\nabla^2, n}^I(s, x) = \frac{4}{R^2} \cos(\theta - \phi) + \sum_{m=2}^{\infty} 4m\frac{\rho^{m-1}}{R^{m+1}} \cos[m(\theta - \phi)], & R > \rho \\ \Theta_{\nabla^2, n}^E(s, x) = \frac{4}{\rho^2} \cos(\theta - \phi) + \sum_{m=2}^{\infty} 4m\frac{R^{m-1}}{\rho^{m+1}} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$M_{\nabla^2, n}(s, x) = \begin{cases} M_{\nabla^2, n}^I(s, x) = \frac{8(v-1)}{R^3} \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} 4m(m+1)(v-1) \frac{\rho^{m-1}}{R^{m+2}} \cos[m(\theta - \phi)], & R > \rho \\ M_{\nabla^2, n}^E(s, x) = - \sum_{m=2}^{\infty} 4m(m-1)(v-1) \frac{R^{m-2}}{\rho^{m+1}} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$V_{\nabla^2, n}(s, x) = \begin{cases} V_{\nabla^2, n}^I(s, x) = \frac{8(v-1)}{R^4} \cos(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} 4m^2(m+1)(v-1) \frac{\rho^{m-1}}{R^{m+3}} \cos[m(\theta - \phi)], & R > \rho \\ V_{\nabla^2, n}^E(s, x) = \sum_{m=2}^{\infty} 4m^2(m-1)(v-1) \frac{R^{m-3}}{\rho^{m+1}} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

A.6. *Tangential derivative of Laplacian of the degenerate kernels*

$$U_{\nabla^2, t}(s, x) = \begin{cases} U_{\nabla^2, t}^I(s, x) = -\frac{4}{R} \sin(\theta - \phi) - \sum_{m=2}^{\infty} 4 \frac{\rho^{m-1}}{R^m} \sin[m(\theta - \phi)], & R > \rho \\ U_{\nabla^2, t}^E(s, x) = -4 \frac{R}{\rho^2} \sin(\theta - \phi) - \sum_{m=2}^{\infty} 4 \frac{R^m}{\rho^{m+1}} \sin[m(\theta - \phi)], & \rho > R \end{cases}$$

$$\Theta_{\nabla^2, t}(s, x) = \begin{cases} \Theta_{\nabla^2, t}^I(s, x) = \frac{4}{R^2} \sin(\theta - \phi) + \sum_{m=2}^{\infty} 4m \frac{\rho^{m-1}}{R^{m+1}} \sin[m(\theta - \phi)], & R > \rho \\ \Theta_{\nabla^2, t}^E(s, x) = -\frac{4}{\rho^2} \sin(\theta - \phi) - \sum_{m=2}^{\infty} 4m \frac{R^{m-1}}{\rho^{m+1}} \cos[m(\theta - \phi)], & \rho > R \end{cases}$$

$$M_{\nabla^2, t}(s, x) = \begin{cases} M_{\nabla^2, t}^I(s, x) = \frac{8(v-1)}{R^3} \sin(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} 4m(m+1)(v-1) \frac{\rho^{m-1}}{R^{m+2}} \sin[m(\theta - \phi)], & R > \rho \\ M_{\nabla^2, t}^E(s, x) = \sum_{m=2}^{\infty} 4m(m-1)(v-1) \frac{R^{m-2}}{\rho^{m+1}} \sin[m(\theta - \phi)], & \rho > R \end{cases}$$

$$V_{\nabla^2, t}(s, x) = \begin{cases} V_{\nabla^2, t}^I(s, x) = \frac{8(v-1)}{R^4} \sin(\theta - \phi) \\ \quad + \sum_{m=2}^{\infty} 4m^2(m+1)(v-1) \frac{\rho^{m-1}}{R^{m+3}} \sin[m(\theta - \phi)], & R > \rho \\ V_{\nabla^2, t}^E(s, x) = - \sum_{m=2}^{\infty} 4m^2(m-1)(v-1) \frac{R^{m-3}}{\rho^{m+1}} \sin[m(\theta - \phi)], & \rho > R \end{cases}$$

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