

# **Detection of damaged components in 3-D frame structures via experimental design**

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# Outline of Presentation

- ◆ **Background**
- ◆ **Theory**
  - **Experimental Design**
  - **Evaluation Function**
- ◆ **Flow of Analysis**
- ◆ **Example Computations**
- ◆ **Concluding Remarks**

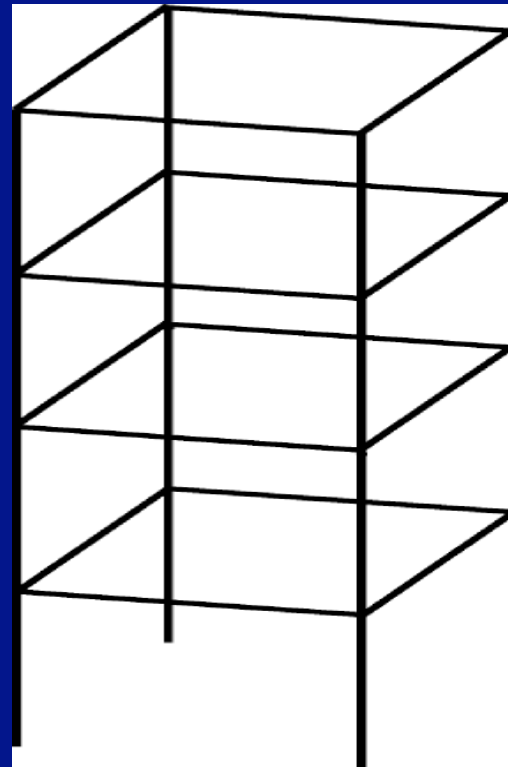
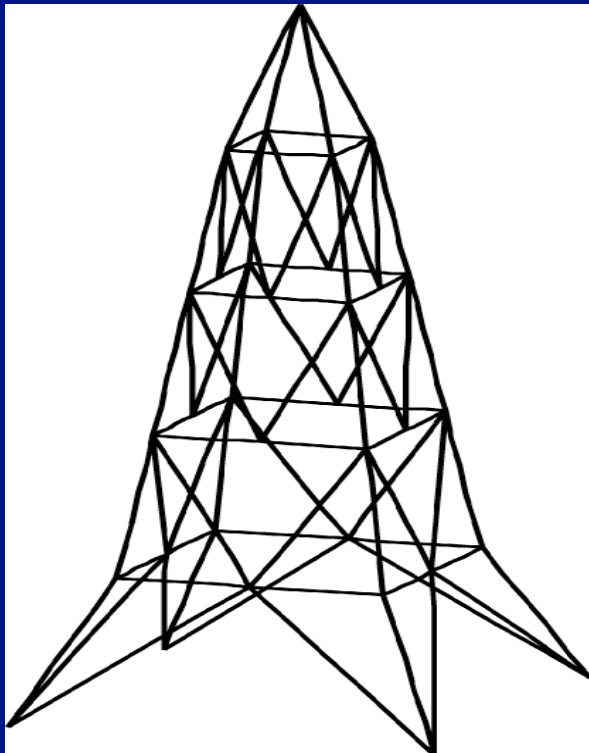
# Background

There is a wide variety of structures in the world.



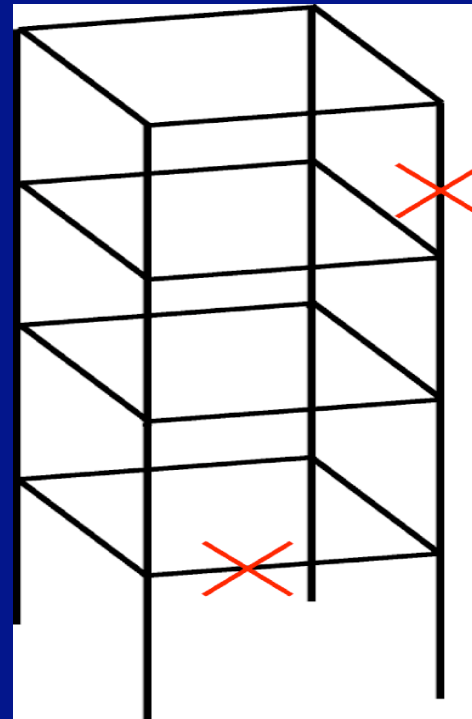
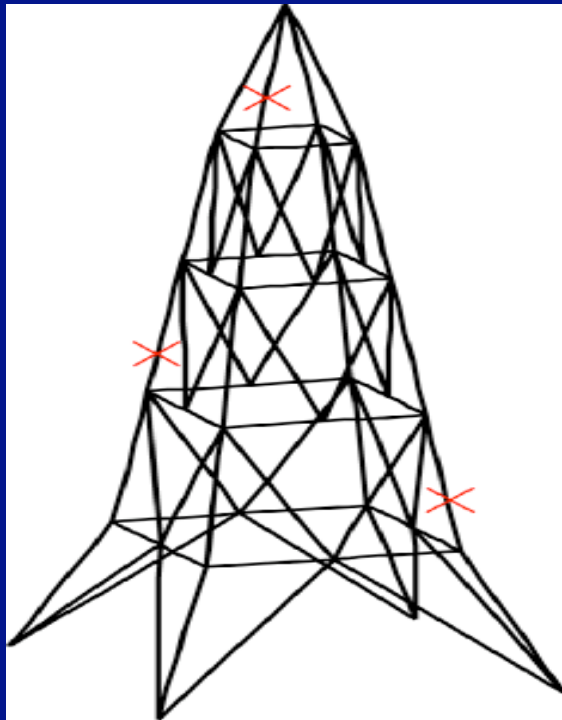
# Background

**Many structures can be modeled as a frame structure.**



# Background

**For health monitoring of the structure, it is important to develop a computer system to identify the damaged components and their damage levels, using the measured data.**



# Background

A computational procedure is available for dynamic displacements of a frame structure.

Displacement responses are different if damaged components and their damage levels are different.



**Sensitivity-based  
Optimization**



Experimental Design  
Combinational Optimization  
Using Orthogonal Table

# Assumptions

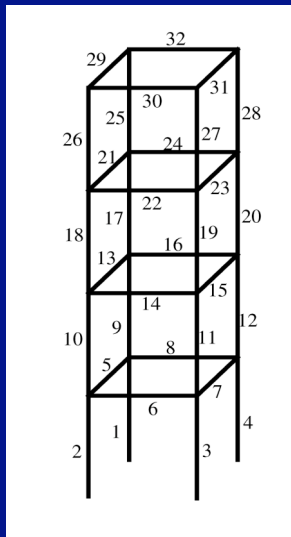
- ◆ Each component of the frame structure is straight and has extensional, bending and torsional rigidities.
- ◆ Damage is interpreted as reduction of Young's modulus.
- ◆ Damage is implemented as three levels of Young's modulus.

# Experimental Design

- ◆ Analysis of each factor's influence on evaluation function using the orthogonal table
- ◆ Analysis of many factors by a smaller number of computations

Factor → Component  
Level → Damage level

**Example:** Analysis of structure composed of 32 components by three levels of damage



Computations for all combinations:

$$3^{32} \doteq 1.85 \times 10^{15}$$

Computations using orthogonal table:

216 to 2268



# Evaluation

Evaluation function  $U_n$  is defined by

$$U_n = \sum_{i=1}^I \sum_{j=1}^J \{ (\bar{u}_{ij} - u_{ij})^2 + (\bar{v}_{ij} - v_{ij})^2 + (\bar{w}_{ij} - w_{ij})^2 \}$$

$j$  : Node     $J$  : Number of nodes

$i$  : Node in time     $I$  : Number of nodes in time

$\bar{u}_{ij}, \bar{v}_{ij}, \bar{w}_{ij}$  : Measured displacements in  $x, y, z$

$u_{ij}, v_{ij}, w_{ij}$  : Computed displacements in  $x, y, z$

$$U_n = 0$$



**Damaged components and their damage levels are identified.**

# Searching Minimal Value of Evaluation Function

		Factors						
		1	2	...	$l$	...	$L$	
Data numbers	1	$k_{11}$	$k_{12}$	...	$k_{1l}$	...	$k_{1L}$	$U_1$
	2	$k_{21}$	$k_{22}$	...	$k_{2l}$	...	$k_{2L}$	$U_2$
	$\vdots$	$\vdots$	$\vdots$		$\vdots$		$\vdots$	$\vdots$
	$n$	$k_{n1}$	$k_{n2}$	...	$k_{nl}$	...	$k_{nL}$	$U_n$
	$\vdots$	$\vdots$	$\vdots$		$\vdots$		$\vdots$	$\vdots$
	$N$	$k_{N1}$	$k_{N2}$	...	$k_{Nl}$	...	$k_{NL}$	$U_N$
	$S_{kl}$	$S_{11}$	$S_{12}$	...	$S_{1l}$	...	$S_{1L}$	
		$\vdots$	$\vdots$		$\vdots$		$\vdots$	
		$S_{k1}$	$S_{k2}$	...	$S_{kl}$	...	$S_{kL}$	
		$\vdots$	$\vdots$		$\vdots$		$\vdots$	
		$S_{K1}$	$S_{K2}$	...	$S_{Kl}$	...	$S_{KL}$	

$S_{kl}$ : Sum of  $U_n$  in factor  $l$  under damage level  $k$

$$S_{kl} = \sum_n U_n(k, l)$$



Search for minimum  $S_{kl}$



Damaged components and their levels are estimated.

# Comparison of Evaluation Function

		Factors						
		1	2	...	$l$	...	$L$	$U_n$
Data numbers	1	$k_{11}$	$k_{12}$	...	$k_{1l}$	...	$k_{1L}$	$U_1$
	2	$k_{21}$	$k_{22}$	...	$k_{2l}$	...	$k_{2L}$	$U_2$
	$\vdots$	$\vdots$	$\vdots$		$\vdots$		$\vdots$	$\vdots$
	$n$	$k_{n1}$	$k_{n2}$	...	$k_{nl}$	...	$k_{nL}$	$U_n$
	$\vdots$	$\vdots$	$\vdots$		$\vdots$		$\vdots$	$\vdots$
	$N$	$k_{N1}$	$k_{N2}$	...	$k_{Nl}$	...	$k_{NL}$	$U_N$
	$S_{kl}$	$S_{11}$	$S_{12}$	...	$S_{1l}$	...	$S_{1L}$	
		$\vdots$	$\vdots$		$\vdots$		$\vdots$	
		$S_{k1}$	$S_{k2}$	...	$S_{kl}$	...	$S_{kL}$	
		$\vdots$	$\vdots$		$\vdots$		$\vdots$	
		$S_{K1}$	$S_{K2}$	...	$S_{Kl}$	...	$S_{KL}$	

Search for minimum  $S_{kl}$



Damage level  $k_{nl}$  is fixed



Search for the structure providing minimum  $U_n$

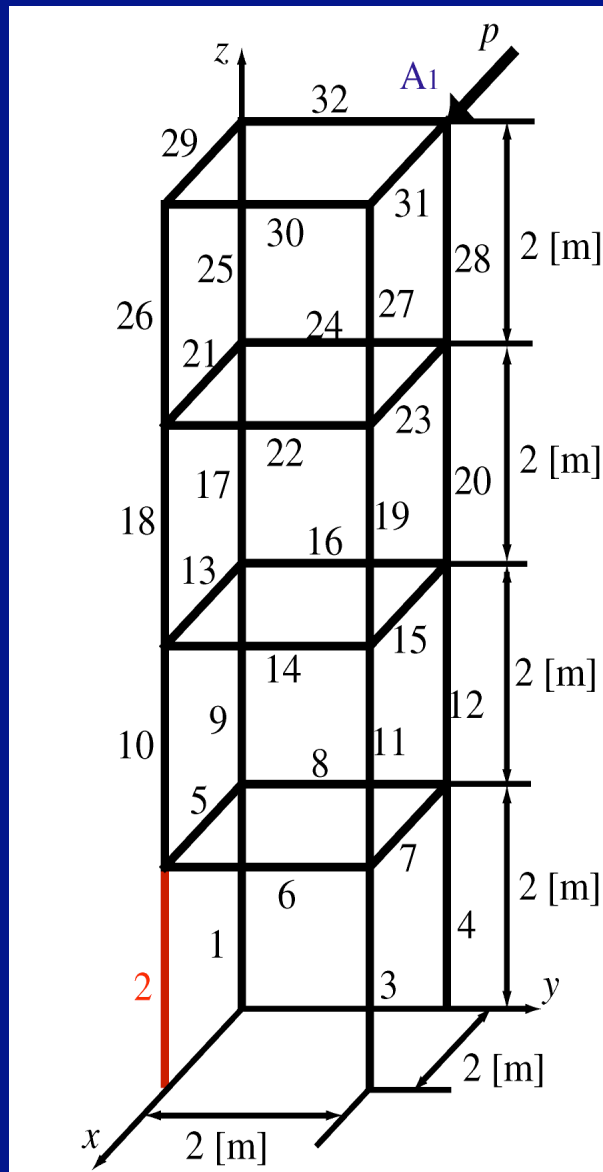


Damaged components and their levels are identified.

# Flow of Analysis

- |        |  |
|--------|--|
| Step 1 | Determine factors and levels based on a priori information |
| Step 2 | Input measured data  |
| Step 3 | Obtain computed data for damaged models of structure       |
| Step 4 | Compute squared sum of measured and computed results       |
| Step 5 | Estimate the damaged components and their damage levels    |
| Step 6 | Carry out Steps 3 to 5 for all components of the structure |
| Step 7 | Carry out Steps 3 to 5 for doubtful components of damage   |
| Step 8 | Iterate Step 7 until the final estimation is obtained      |
| Step 9 | Output the final results                                   |

# Example 1



Member 2 : Young's modulus 50%

Structure:

Bottom is clamped to the plane  $xy$   
Each member has the same circular cross-section with radius 0.01[m]

Material constants:

Young's modulus  $E = 210$  [GPa]

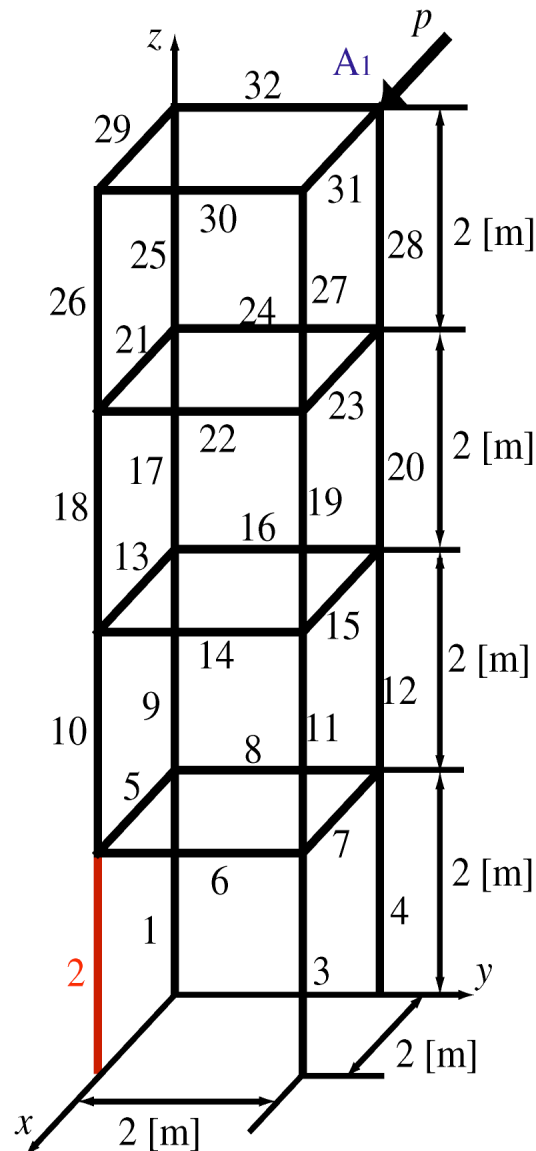
Density  $\rho = 7860$  [kg/m<sup>3</sup>]

Poisson's ratio  $\nu = 0.3$

Concentrated load  $P = 100H(t)$  [N] is applied to point  $A_1$  along the axis  $x$  for 0.5 [s]. The displacements in  $x$  and  $y$  directions are measured for 2.0[s] at equal 10 steps.

# Example 1

Member 2 : Young's modulus 50%

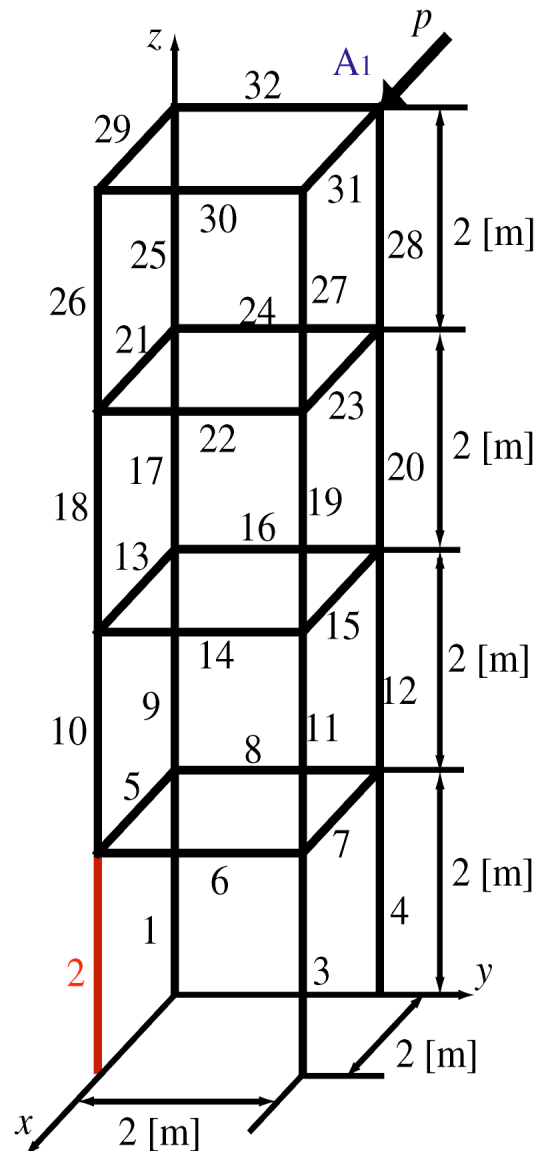


Orthogonal Table

	Member 1	Member 2	Member 3	Member 4	$U_n$
No.1	100%	100%	100%	100%	3.387E- 04
No.2	100%	50%	50%	50%	1.687E- 02
No.3	100%	25%	25%	25%	7.888E- 02
No.4	50%	100%	50%	25%	5.093E- 02
No.5	50%	50%	25%	100%	9.439E- 03
No.6	50%	25%	100%	50%	4.765E- 03
No.7	25%	100%	25%	50%	4.926E- 02
No.8	25%	50%	100%	25%	1.943E- 02
No.9	25%	25%	50%	100%	5.910E- 03
$S_{1l}$	9.608E- 02	1.005E- 01	2.454E- 02	1.569E- 02	
$S_{2l}$	6.514E- 02	4.574E- 02	7.371E- 02	7.089E- 02	
$S_{3l}$	7.460E- 02	8.955E- 02	1.376E- 01	1.492E- 01	

# Example 1

Member 2 : Young's modulus 50%

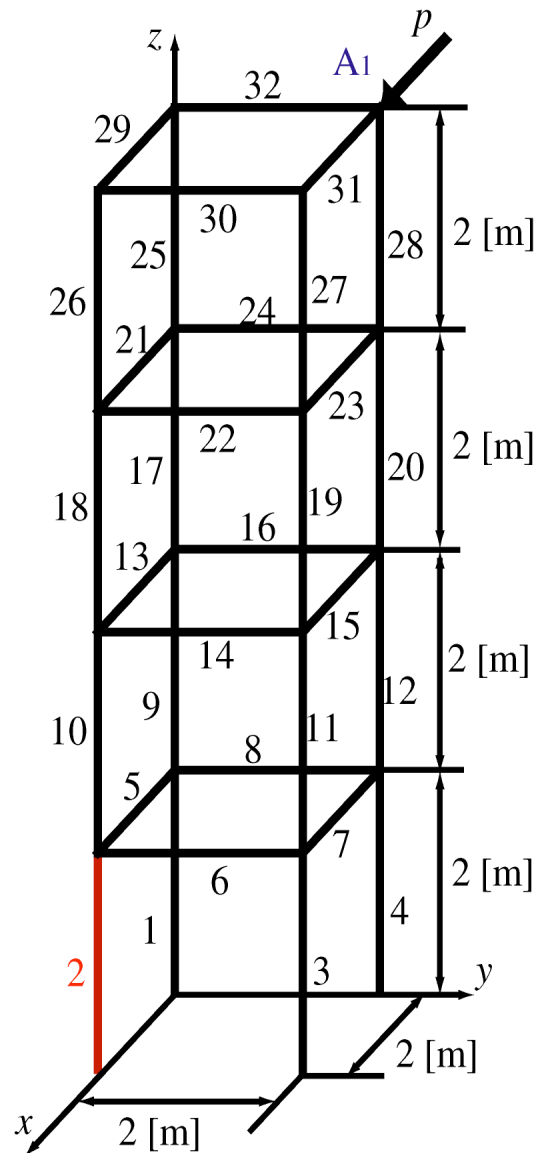


## Orthogonal Table

	Member 1	Member 2	Member 3	Member 4	$U_n$
No.1	100%	100%	100%	100%	3.387E- 04
No.2	100%	50%	50%	100%	2.620E- 03
No.3	100%	25%	25%	100%	9.183E- 03
No.4	50%	100%	50%	100%	3.751E- 03
No.5	50%	50%	25%	100%	9.439E- 03
No.6	50%	25%	100%	100%	8.764E- 04
No.7	25%	100%	25%	100%	1.192E- 02
No.8	25%	50%	100%	100%	7.720E- 04
No.9	25%	25%	50%	100%	5.910E- 03
$S_{1l}$	1.214E- 02	1.601E- 02	1.987E- 03		
$S_{2l}$	1.407E- 02	1.283E- 02	1.228E- 02		
$S_{3l}$	1.861E- 02	1.597E- 02	3.055E- 02		

# Example 1

Member 2 : Young's modulus 50%



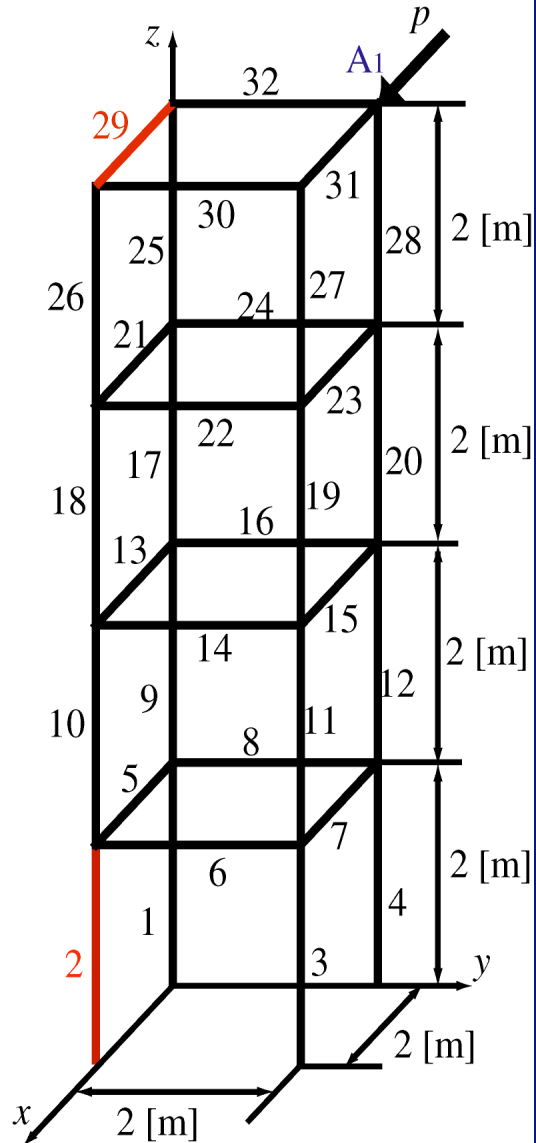
## Orthogonal Table

	Member 1	Member 2	Member 3	Member 4	$U_n$
No.1	100%	100%	100%	100%	3.387E- 04
No.2	100%	50%	100%	100%	7.035E- 07
No.3	100%	25%	100%	100%	2.202E- 04
No.4	50%	100%	100%	100%	5.234E- 04
No.5	50%	50%	100%	100%	2.307E- 04
No.6	50%	25%	100%	100%	8.764E- 04
No.7	25%	100%	100%	100%	8.417E- 04
No.8	25%	50%	100%	100%	7.720E- 04
No.9	25%	25%	100%	100%	4.872E- 02
$S_{1I}$	5.595E- 04	1.704E- 03			
$S_{2I}$	1.630E- 03	1.003E- 03			
$S_{3I}$	3.609E- 03	3.092E- 03			



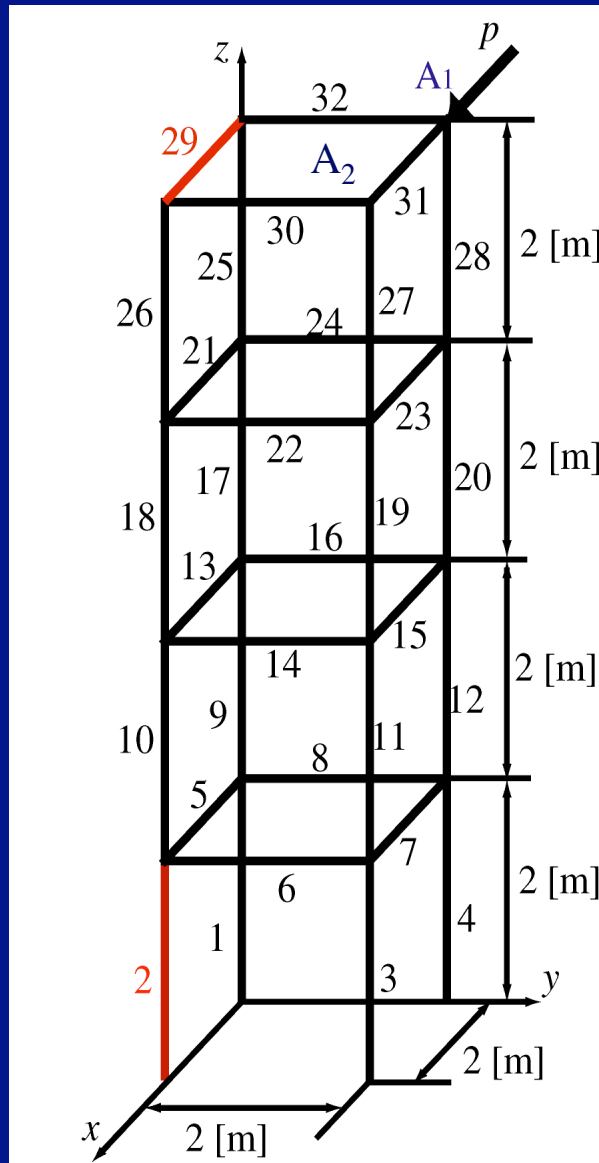
## Example 2

Member 2 : Young's modulus 50%  
Member 29 : Young's modulus 25%



Member	Exact	Analysis	Member	Exact	Analysis
1	100 %	25 %	17	100 %	25 %
2	50 %	25 %	18	100 %	100 %
3	100 %	100 %	19	100 %	100 %
4	100 %	100 %	20	100 %	100 %
5	100 %	50 %	21	100 %	50 %
6	100 %	100 %	22	100 %	100 %
7	100 %	100 %	23	100 %	100 %
8	100 %	100 %	24	100 %	100 %
9	100 %	100 %	25	100 %	25 %
10	100 %	100 %	26	100 %	25 %
11	100 %	100 %	27	100 %	100 %
12	100 %	50 %	28	100 %	100 %
13	100 %	100 %	29	25 %	25 %
14	100 %	100 %	30	100 %	100 %
15	100 %	100 %	31	100 %	100 %
16	100 %	100 %	32	100 %	100 %

## Example 2



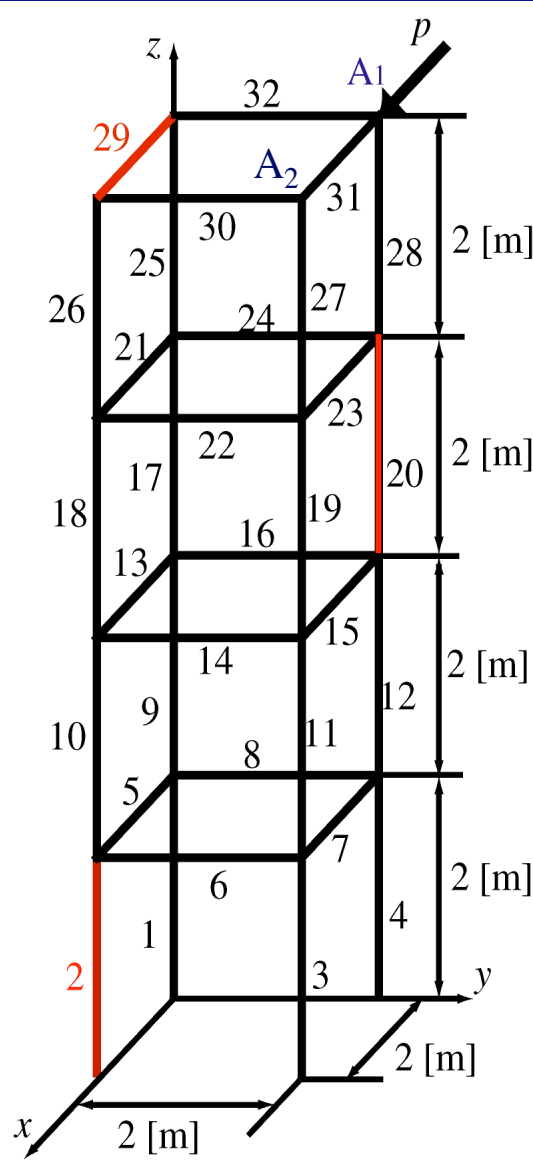
### Final results

Member 2 : Young's modulus 50%

Member 29 : Young's modulus 25%

Doubtful components are again checked using the orthogonal table. Finally, it is estimated that the components 2 and 29 are damaged as into the assumed levels.

### Example 3 Structure with damage different from the assumed damage level



Member 2 : Young's modulus 20%  
Member 20 : Young's modulus 40%  
Member 29 : Young's modulus 60%

Experimental design assumes the three levels 25%, 50% and 100%.  
We want to know what happens in such a case.

## Example 3

Member 2 : Young's modulus 20%  
Member 20 : Young's modulus 40%  
Member 29 : Young's modulus 60%

1st trial (Node  $A_1$ ):

Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %
1	100	9	100	17	100	25	100
2	25	10	100	18	100	26	100
3	100	11	100	19	100	27	100
4	100	12	100	20	25	28	100
5	100	13	100	21	100	29	25
6	100	14	100	22	100	30	100
7	100	15	100	23	100	31	100
8	100	16	100	24	100	32	100

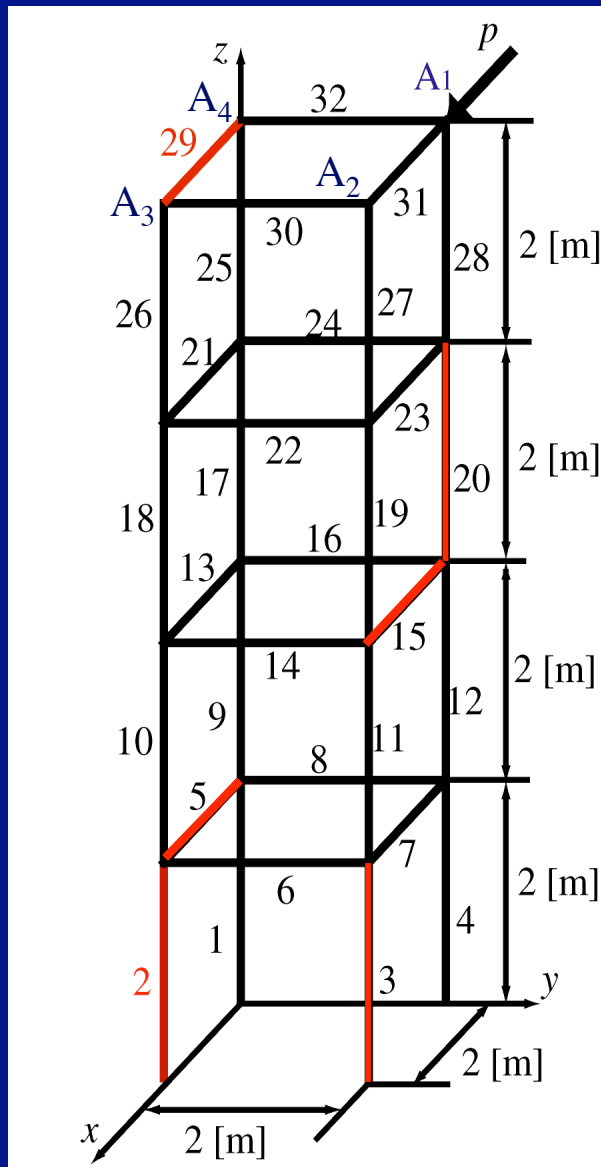
## Example 3

Member 2 : Young's modulus 20%  
 Member 20 : Young's modulus 40%  
 Member 29 : Young's modulus 60%

2nd trial (Nodes  $A_1$  and  $A_2$ ):

Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %
1	100	9	100	17	100	25	100
2	25	10	100	18	100	26	100
3	100	11	100	19	100	27	100
4	100	12	100	20	50	28	100
5	100	13	100	21	100	29	50
6	100	14	100	22	100	30	100
7	100	15	100	23	100	31	100
8	100	16	100	24	100	32	100

## Example 4 Structure with several damaged members with different levels from the assumed ones



Member 2 : Young's modulus 50%  
 Member 3 : Young's modulus 25%  
 Member 5 : Young's modulus 55%  
 Member 15 : Young's modulus 35%  
 Member 20 : Young's modulus 20%  
 Member 29 : Young's modulus 50%

Experimental design assumes the three levels 25%, 50% and 100%. We want to know what happens in this case.

## Example 4

Member 2 : Young's modulus 50%  
 Member 3 : Young's modulus 25%  
 Member 5 : Young's modulus 55%  
 Member 15 : Young's modulus 35%  
 Member 20 : Young's modulus 20%  
 Member 29 : Young's modulus 50%

1st trial (Node  $A_1$ ):

Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %
1	100	9	100	17	100	25	100
2	50	10	100	18	100	26	100
3	50	11	100	19	100	27	100
4	100	12	100	20	25	28	100
5	50	13	100	21	100	29	50
6	100	14	100	22	100	30	100
7	100	15	25	23	100	31	100
8	100	16	100	24	100	32	100

## Example 4

Member 2 : Young's modulus 50%  
 Member 3 : Young's modulus 25%  
 Member 5 : Young's modulus 55%  
 Member 15 : Young's modulus 35%  
 Member 20 : Young's modulus 20%  
 Member 29 : Young's modulus 50%

2nd trial (Nodes  $A_1$  to  $A_4$ ):

Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %	Rod No.	Rigidity in %
1	100	9	100	17	100	25	100
2	50	10	100	18	100	26	100
3	25	11	100	19	100	27	100
4	100	12	100	20	25	28	100
5	50	13	100	21	100	29	50
6	100	14	100	22	100	30	100
7	100	15	25	23	100	31	100
8	100	16	100	24	100	32	100



# Concluding Remarks

- ◆ Experimental design is very tough and robust for damage detection in frame structures.
- ◆ Damage detection can be done with a fewer number of points for measurement.
- ◆ Based on the estimated results by the present method we may improve the solutions via the sensitivity-based inverse analysis.